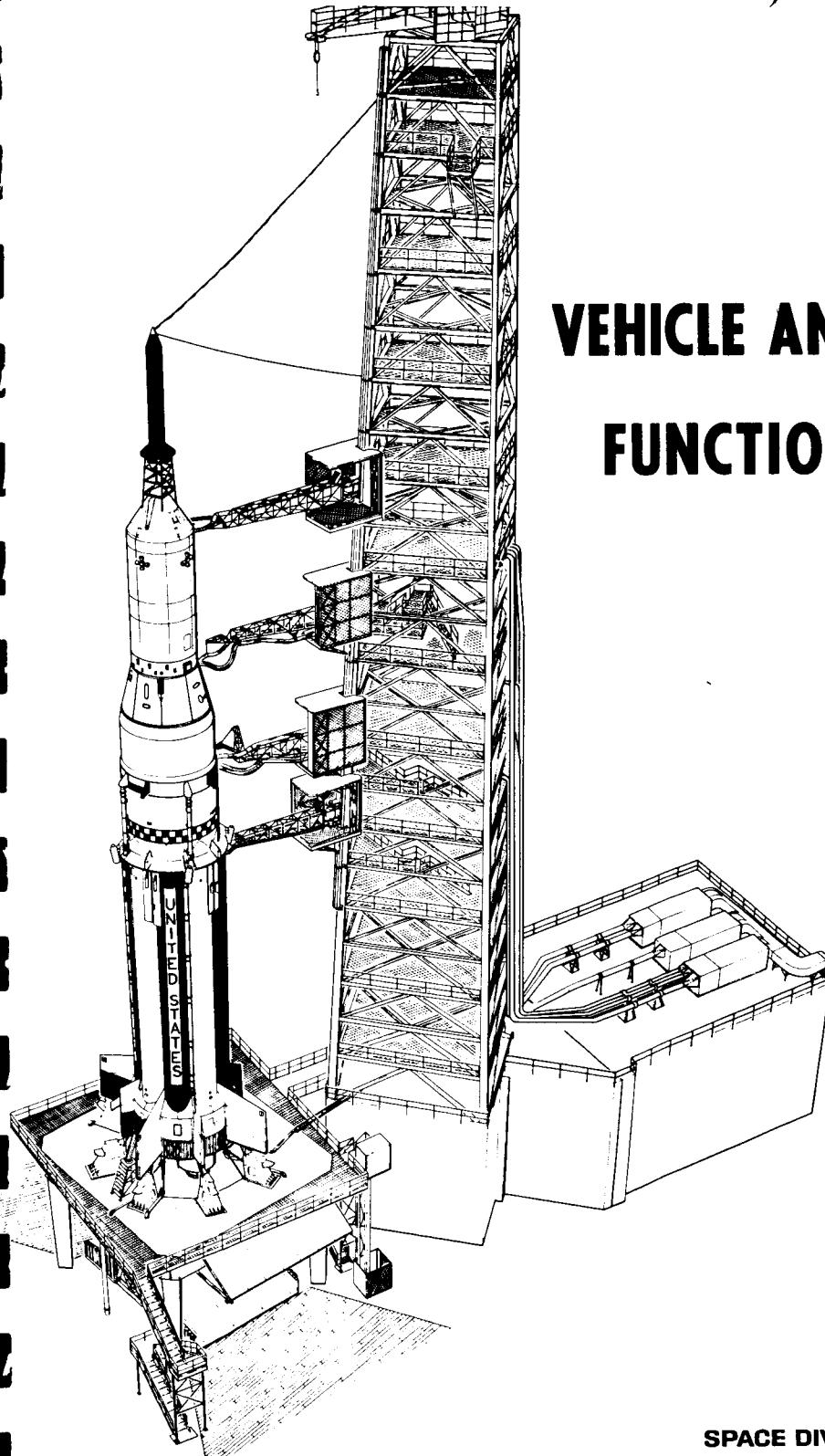


N65-18115

(ACCESSION NUMBER)

67
(PAGES)CR 57067
(NASA CR OR TMX OR AD NUMBER)

(THRU)	1
(CODE)	II
(CATEGORY)	II

HEC-D042**VOLUME II****SA-9**

VEHICLE AND LAUNCH COMPLEX FUNCTIONAL DESCRIPTION

LOX SYSTEM

GPO PRICE \$ _____

OTS PRICE(S) \$ _____

Hard copy (HC) 3.00Microfiche (MF) .75

SPACE DIVISION  **CHRYSLER**
CORPORATION

HUNTSVILLE OPERATIONS

**HEC-D042
VOLUME II**

SA-9

**VEHICLE AND LAUNCH COMPLEX
FUNCTIONAL DESCRIPTION**

LOX SYSTEM

APRIL 1964

ENGINEERING COMMUNICATIONS DEPARTMENT



FOREWORD

This volume has been prepared for the Functional Integration Section, Systems Integration and Operations Branch, Vehicle Systems Division, Propulsion and Vehicle Engineering Laboratory, by the Engineering Communications Department, Chrysler Corporation Space Division, under contract number NAS8-4016.

The following series, of which this volume is a part, functionally describes the mechanical and electromechanical systems of Saturn I, SA-9 space vehicle and Launch Complex 37:

- Volume I. RP-1 Fuel System
- Volume II. LOX System
- Volume III. LH₂ System
- Volume IV. Nitrogen and Helium Storage Facility
- Volume V. Pneumatic Distribution System
- Volume VI. Environmental Control System
- Volume VII. Launch Pad Accessories
- Volume VIII. H-1 Engine and Hydraulic System
- Volume IX. RL10A-3 Engine and Hydraulic System
- Volume X. Separation and Flight Termination Systems
- Volume XI. Supplement: Legend and Composite Schematic

Each volume contains mechanical schematics and a list of applicable finding numbers.

Volume II describes those components that are active during countdown, launch, and flight: it specifically excludes maintenance procedures. Only information available by March 1, 1964, has been included.

TABLE OF CONTENTS

	Page
1. DESCRIPTION	1
1.1. Storage Facility	1
1.2. Launch Control Center.	2
1.3. Launch Pad B	4
1.4. S-I Stage.	6
1.5. S-IV Stage	6
2. OPERATION.	7
2.1. Storage Facility	7
2.2. Storage Facility, Launcher, LOX Pit, and S-I Stage	10
2.3. Launcher, LOX Pit, and S-I Stage	13
2.4. Storage Facility, S-IV Tower Complex, and S-IV Stage	14
2.5. S-IV Tower Complex and S-IV Stage.	16
2.6. Flight	18
LIST OF FINDING NUMBERS.	21
MECHANICAL SCHEMATICS.	63
DISTRIBUTION	69

1. DESCRIPTION

The Launch Complex 37 LOX system fills, replenishes, pressurizes, and drains the S-I and S-IV stage LOX containers on Saturn SA-9. The system consists basically of a LOX storage facility approximately 740 feet from the launch pad and of LOX loading equipment in the launch control center (LCC), in the automatic ground control station (AGCS), in the LOX pit, and on launch pad B.

Two independent systems on SA-9 store and supply LOX for engine operation: one on the S-I stage (H-1 engine), the other on the S-IV stage (RL10A-3 engine).

After the LOX storage tanks have been filled and the system checked out, LOX is transferred to the vehicle through an automatic sequence that begins with storage tank pressurization. The fill and drain lines are precooled prior to the fill operations and the replenish lines are precooled prior to the replenish operations. This cooling minimizes pressure buildups and prevents improper LOX flow. Replenish operations start at the completion of main fill operations, approximately two hours before ignition, and terminate at T -30 seconds. Replenish compensates for LOX loss through vaporization and tops off the LOX containers.

Drain operations return LOX to the storage facility at the conclusion of checkout procedures and after an abort.

The master controls for LOX loading are in the LCC. The LOX control panel in the LCC and the LOX tanking computer in the AGCS are the control points for S-I stage LOX loading. Controls and indicators for the S-IV Propellant Utilization (PU) System, provided by Douglas Aircraft Company (DAC) for S-IV stage LOX loading, are an integral part of the Launch Complex 37 LOX system. Since NASA document LTM-4-12 prepared by K-DP describes and illustrates the control panels, and since the LOX transfer is basically an automatic operation, the description of indicators and manual controls in this volume is cursory. Only the information needed to understand the basic system operation is included.

1.1. Storage Facility

A main storage tank, sensing panel, and vaporization (tank pressurization) system; a replenish tank, sensing panel, and vaporization (tank pressurization) system; a pneumatic control panel; and transfer lines (some with filters, pumps, valves, and associated components) comprise the LC37 LOX storage facility.

The main storage tank holds 125,000 gallons of LOX with a 12-percent ullage. The 4 1/2 feet of annular space between the steel outer wall and the aluminum inner wall of the spherical tank is filled with perlite insulation.

The replenish tank holds 28,000 gallons of LOX with a five-percent ullage. The one foot of annular space between the steel outer wall and the aluminum inner wall of the cylindrical tank is filled with perlite insulation.

The vaporization systems, one for the main tank and one for the replenish tank, convert LOX to GOX, which pressurizes the tanks' ullage during LOX loading. The main and replenish tank sensing panels have (a) gages that indicate LOX quantity, ullage pressure, and replenish tank annular space vacuum (main tank annular space pressure is indicated by a gage on the main tank); (b) a valve that vents the ullage during tank filling, allows LOX to flow out a line for visual indication when the tank is full, and is then closed to allow ullage pressurization; (c) pressure switches and transducers that provide LOX tank quantity and pressure signals to indicators in the LCC. The panels also contain pressure switches that prevent overpressurization of the tanks, and the main tank panel contains pressure switches that illuminate indicators in the LCC when S-I and S-IV LOX fill pump discharge pressures are adequate.

The pneumatic control console filters and reduces the pressure of GN₂ in four steps for actuation of LOX system valves and regulators and for purging LOX lines.

1.2. Launch Control Center

The master controls for LOX loading operations are in propellant loading racks numbers 3, 4, 5, and 6 in the LCC. Equipment in these racks initiates and controls LOX transfer operations, displays LOX system conditions, allows manual checkout of components, and initiates and controls simulated LOX transfer. Rack number 3 houses a propellant monitor, sequence recorder readout, component readout distributor, and a computer distributor. Rack number 4 houses a manometer monitor, sequence recorder readout, and a relay distributor. Rack number 5 houses a LOX dc power panel, LOX components panel, LOX control panel, and a relay distributor. Rack number 6 houses the LOX supply monitor, digital indicator, LOX computer, and a relay distributor.

1.2.1. LOX Components Panel. Component condition indicators (lights) and manual controls (toggle switches) for remotely operated LOX system components are on the LOX components panel. The replenish and throttle control valve display consists of two green lights and one white light. Each other component condition display consists of one green or red light and one white light. Nomenclature on the white light indicates the normal or deenergized condition. Nomenclature on the green light indicates the operated position, and, on the red light, indicates the over-pressure or over-fill position. Use of the switches to manually control the components is dependent on the mode selector switch setting on the LOX control panel.

1.2.2. LOX Control Panel. Switches on this panel initiate the automatic filling and draining operations, initiate the simulated LOX transfer operation, and transfer control (manual) to the LOX components panel. System condition indicators (lights similar to those on the LOX components panel), a power switch, and a 'return S-IV control' switch are also on this panel. Indicator

Table 1. Switches and Functions, LOX Control Panel

Switch	Function
Sequence	Selects automatic or semi-automatic mode. The 'auto' position initiates and automatically sequences all LOX transfer operations through completion of S-IV and S-I LOX replenish. The 'semi-auto' position transfers control of LOX transfer operations to five pushbutton switches for semi-automatic sequencing through completion of S-I main LOX fill operation. The pushbutton switches must be operated in sequence; however, the operator can revert to any previous operation (except S-IV main fill).
Pressurize push-button	Reverts the LOX filling system to 'pressurized storage tanks' condition.
Precool push-button	Initiates the S-IV stage LOX fill line precool operation.
Main fill S-IV pushbutton	Initiates the S-IV stage main LOX fill operation.
Precool S-I push-button	Initiates the S-I LOX fill line precool operation.
Main fill S-I pushbutton	Initiates the S-I stage main LOX fill operation. The remaining LOX transfer operations to the vehicle are automatically sequenced by signals received from the S-I tanking computer and the S-IV propellant utilization system.
Ready for final replenish	Indicates on firing panel that the S-I stage is LOX-loaded to $100 \pm 0.10\%$ and that the S-IV stage is ready for final replenishing.
Power	Controls power for entire LOX system.
S-IV drain	Initiates LOX drain from the S-IV stage LOX container.
S-I drain	Initiates LOX drain from the S-I stage LOX containers.
Control return	Returns control of S-IV stage umbilical tower and vehicle valves to DAC.
Function (mode) selector	Selects one of three modes of operation and off. 'Operate' provides automatic or semi-automatic sequential operation. 'Simulate' is the same as 'operate,' except pump motors do not operate and there is no LOX flow (manual valves remain closed). 'Manual' transfers control to LOX Components Panel for manual operation of remotely operated components.

lights across the top of the panel sum up all necessary prerequisites for LOX filling and draining. Table 1 lists the switches and their functions.

1.2.3. Computer Distributor Panel. This panel, in propellant loading rack No. 3, distributes signals from the AGCS and the LOX storage facility to the LOX supply monitor, LOX tanking computer panel, RP-1 density computer panel, and RP-1 tanking computer panel.

1.2.4. LOX Computer Panel. This panel provides the controls and indicators for the LOX loading computer system. Two readouts are provided: one displays the precalibrated nominal differential set pressure in psi; the other displays the dialed-in correction to the set pressure in psi and its polarity. A telephone-type dial, a polarity switch, and a 'clear' pushbutton are used to effect pressure corrections within ± 0.46 psi.

A power switch applies all power to the computer system. Fuses are provided on the panel front only for the power used in the panel. Indicators (lights) for '75% full' and 'stop fast fill' illuminate when energized either from the computer or from manual initiation switches below the lights. A 'computer test' indicator is illuminated when the computer system is in the test sequence. A switch below this light manually initiates the test sequence. A 'replenish rate' potentiometer is used to adjust the LOX replenish valve control signal as necessary to provide LOX to compensate for boiloff.

1.2.5. Digital Indicator Panel. This panel indicates the percent LOX level in the S-I stage LOX containers. The percent displayed is based on the algebraic sum of the precalibrated LOX loading computer nominal set point and the set point correction dialed in.

The digital indicator panel receives inputs from the data transmitter at the LOX loading computer in the AGCS. This information is translated and displayed in decimal form. For further information concerning LOX loading panels in the LCC, refer to LTM-4-12.

1.3. Launch Pad B

1.3.1. Automatic Ground Control Station. LOX loading equipment in the AGCS consists of panels mounted in the propellant level control system data transmission rack, the propellant level control system calibration and monitor rack, and the launcher accessories rack number 2 - all on the second floor.

1.3.1.1. LOX Loading Computer. The LOX loading computer, in the AGCS propellant level control system calibration and monitor rack, generates an electrical analog of sensed differential pressure. The electrical analog and a reference voltage are presented by the computer to its ratiometer for readout. The sensed differential pressure, proportional to LOX quantity, is brought in through sensing lines from the S-I stage center LOX container.

The computer generates switching points for 75 percent full and stop fast fill, which are monitored from the LOX computer panel indicators in the LCC. (See 1.2.4.). The computer also generates signals that proportionally control the LOX replenish valve.

1.3.1.2. Ratiometer-Data Transmitter. The ratiometer-data transmitter, located in the AGCS propellant level control system data transmission rack, is composed of three modules. A reference voltage and a signal voltage received from the LOX loading computer are compared in the ratiometer and displayed in percent on an indicator. The indication is based on the algebraic sum of the precalibrated LOX loading computer nominal set point and the set point correction dialed in. This indication is electrically transmitted to the LCC where it is displayed on the digital indicator (paragraph 1.2.5.).

1.3.1.3. LOX Level Manometer. The LOX level manometer, in the AGCS manometer rack, is a closed, non-adjustable, cistern-type, mercury pressure gage with a range of 0.0 to 30.6 psi. The manometer senses differential pressure across the LOX loading computer and displays the pressure with an accuracy of 0.02 percent. The differential pressure represented by the mercury column height is automatically sensed and converted into a synchro transmitter position, binary coded decimal, and a digital indication. Visual displays of column height appear in pounds per square inch. The manometer is remotely controlled from either the propellant monitor panel in the LCC or from the propellant calibration panel in the propellant level control system calibration and monitor rack in the AGCS.

1.3.1.4. Propellant Calibration Panel. This panel, located in the AGCS propellant level control system calibration and monitor rack, controls the LOX level manometer and repeats on a counter the manometer sensed differential pressure. The panel also provides a power distribution point for all calibration and monitor system units and assemblies.

For further information concerning AGCS LOX Loading Control, refer to LTM-4-16 and LTM-4-19.

1.3.2. S-IV Tower Complex. LOX loading equipment on the tower includes the main fill and topping control valve complex; the S-IV fill, replenish, and drain line; the S-I and S-IV fill and replenish precool valves; and associated solenoid control valves and plumbing. The main fill and topping control valve complex is installed as a unit at the umbilical tower 108-foot level. The S-IV fill, drain, and replenish line connects the valve complex to the S-IV stage umbilical housing through swing arm number 2 (described in Volume VII). Two other LOX transfer lines connect to the valve complex: one, the S-IV replenish line, from the S-I replenish line; the other, the S-IV fill and drain line, from the main storage tank.

1.3.3. Launcher and LOX Pit. LOX loading equipment on the launcher consists of the S-I fill and drain line, which connects to the S-I stage fill and drain nozzle through the LOX mast as described in Volume VII; the S-I LOX replenish coupling-half (mounted on the launcher), which connects to the S-I stage at fin position IV; pressure sensing lines from the center LOX container to the computer in the AGCS, which connect through coupling-halves on short cable mast number 4 (Volume VII); the LOX container prepressurization lines, the fill and drain valve control pressure line, the replenishing valve opening control pressure line, and the LOX bubbling line, which also connect to the S-I stage through coupling-halves on mast number 4; the main fill and

replenish solenoid control valves, LOX container pressurization solenoid control valves, and the LOX bubbling solenoid control valve.

LOX loading equipment located in the LOX pit near the launcher consists of the computer controlled pneumatic relay, replenish throttle control valve, replenish control solenoid, S-I throttle bypass valve, replenish line drain valve, S-I mast drain valve, and the S-I main fill valve.

1.4. S-I Stage

Five cylindrical containers store LOX and, during S-I stage powered flight, supply it to the H-1 engines. The four 70-inch diameter containers (0-1, 0-2, 0-3, and 0-4) are installed 90 degrees apart, between fuel containers, forming a circular pattern around the 105-inch diameter center LOX container (0-C). The LOX containers also provide the structural support between the thrust structure and the spider beam. Each outer LOX container has a volume of 11,000 gallons; the center container has a volume of 24,000 gallons. Four interconnect lines join the outer container ullages to the GOX pressurization manifold, which attaches to the center container forward bulkhead. These lines equalize the ullage pressure in the five containers and allow pressurization through the GOX diffuser line that enters the center container from a manifold to the H-1 engine heat exchangers. Four lines connect the outer container sumps to the center container sump; this allows fill through a single coupling, replenish through a single coupling, and provides equal LOX levels during flight.

Two vent and relief valves and one vent valve provide overpressurization protection for the LOX containers. LOX converted to GOX pressurizes the LOX containers during flight; helium prepressurizes the containers from T -100 seconds until mainstage operation commences.

Eight LOX suction lines, with flexible bellows joints, supply LOX to the engines through prevalves held open by GN₂ control pressure. Two lines from each outer container sump lead to the engine turbopumps: one on an outboard engine, the other on an inboard engine.

1.5. S-IV Stage

The S-IV LOX system stores and supplies pressurized LOX to the six RL10A-3 engines during S-IV powered flight. The system consists of a 1263-cu ft container; six thermally insulated suction lines with flexible bellows joints; a capacitance-type LOX level sensor; a cold helium system, which supplies high pressure helium for container pressurization; a LOX bubbling system; and associated valves, switches, orifices, and plumbing.

The S-IV umbilical plate provides all necessary connections to GSE through swing arm number 2 for S-IV LOX system operation prior to liftoff.

2. OPERATION

Components required for various sequential operations are grouped in this section according to location.

Sequences leading to vehicle launch are described under the area or areas that contain the components required for each sequence. The storage facility, launcher, LOX pit, S-IV tower complex, S-I stage, and S-IV stage comprise the locational areas. Only the information needed to understand the basic system operation is included.

Figures 1, 2, and 3 at the back of this document represent the LOX system and should be used in conjunction with the text.

2.1. Storage Facility

Equipment located in the storage facility receives, stores, and transfers LOX to and from the launch pad.

2.1.1. Pneumatic Control Console (PCC). The PCC receives 3500-psig GN₂ from the nitrogen and helium storage facility; drops the pressure to 750, then 120, and then 25 psig; and also drops the pressure to 50 psig in one step (figure 2). Pneumatic Filter A2707 removes contaminants from the GN₂, thereby providing regulator contamination protection. Orifice A2734 reduces 3500 psig to 50 psig; Regulator A2704 reduces 3500 psig to 750 psig; Regulator A2705 reduces 750 psig to 120 psig; and Regulator A2706 reduces 120 psig to 25 psig. Three outlets supply GN₂ to other parts of the LOX system: one at 750, one at 50, and one at 25 psig.

Pressure Gages A2700, A2701, A2702, and A2703 provide pressure indications within each system leg. Pressure Switch A2725 sends a 750-psig OK signal to the LOX control panel in the LCC; Pressure Switches A2711 and A2725 send 750 and 25-psig OK signals to the storage facility pneumatic control cabinet panel. Relief Valves A2708, A2709, A2710, and A2735 provide automatic individual leg relief at 900, 120, 35, and 55 psig, respectively. Manual Valves A2712, A2713, A2714, A2720, A2726, and A2715 allow isolation of various segments of the system. Manual Valves A2718, A2721, A2723, A2724, A2722, A2719, A2717, A2730, and A2716 provide complete system manual venting. (Manual Valve A2771 can be closed to isolate the entire console for maintenance.)

2.1.2. Main Tank Fill. LOX, transported to Complex 37 in trailers, is pumped into Main Storage Tank A300 through hoses that couple to the main tank fill coupling-half assemblies (figure 3).

LOX passes from the trailers through Coupling-Half Assemblies A108, A109, A120, A121, and A122; through Manual Valves A91, A67, A116, A117, and A118; past Relief Valve A80, which relieves at 75 psig; through Strainer A126 and Manual Valve A308; and into the main tank. When the LOX reaches the desired

level, it flows through Trycock Valve A311. The trycock valve is then closed and LOX tank fill terminates. Manual Valve A316 is opened to allow the tank LOX level to be read on Differential Pressure Gage A312. The main tank inlet and outlet lines drain through Manual Valves A127, A128, A129, A130, A131, A113, A68, A70, A71, and A112; into a common line; and through Check Valve A114.

2.1.3. Main Tank Annular Space Pressurization. The annular space between the tank walls receives dry GN₂ at 0.1 psig to prevent moisture accumulation in the perlite insulation. Either Storage Cylinder Assembly A339 or A340 supplies the GN₂ while the other remains on standby. When Cylinder A340 supplies GN₂, Manual Valve A334 remains closed. 2000-psig GN₂ from Cylinder A340 is reduced to 50 psig by Regulator A336 and further reduced to 0.1 psig by Regulator A331 before entering the main tank annular space. With Cylinder A340 on standby (Manual Valve A333 closed), 2000-psig GN₂ from Cylinder A339 is reduced to 50 psig by Regulator A335 before being reduced to 0.1 psig by Regulator A331 for annular space pressurization. Pressure is displayed on Gages A337, A338, and A344. Pressure and Vacuum Relief Valve A342 maintains a constant GN₂ bleed from the annular space.

2.1.4. Main Tank Pressurization. A vaporization system, which converts LOX to GOX, pressurizes the ullage of Main Tank A300 to 30 psig. This ullage pressure minimizes boiloff losses and assists LOX flow from the tank. The sequence switch on the LCC LOX control panel is placed in the automatic position to automatically sequence all LOX transfer operations through completion of S-IV and S-I LOX replenish.

Closure of Valves A309, A307, A308, and A310 precedes main tank pressurization; then Manual Valve A323 is opened to allow LOX to flow from the tank to Heat Exchanger A306 for main tank pressurization. Dc power to Solenoid Valves A2736 and A2737 allows 750-psig GN₂ to close normally open Main Tank Vent Valve A1. (Manual Valve A327 provides a means of preventing main tank pressure leakage should Valve A1 fail to close.) Power on Valves A2738 and A2739 allows control pressure to open normally closed Main Tank Pressurization Valve A301. LOX flows from the tank through Manual Valve A323, Valve A301, and Flow Control Valve A306 which is held in the fully open position by 25-psig GN₂ control pressure. From Valve A306, LOX flows through Heat Exchanger A305, where it is converted to GOX by ambient air being blown across the heat exchanger coils by two motor-operated fans. The GOX passes through the return line to pressurize the tank's ullage.

A line leading to Controller A328 transmits ullage pressure, which varies the controller's outlet pressure from 3 to 15 psig, thus varying the opening through Valve A306 proportionally. In this way, a relatively constant ullage pressure is maintained. Should the ullage pressure exceed 30 psig, Pressure Switch A447 in the main tank sensing panel would actuate, taking power off Valves A2736 and A2737. This action allows Valve A1 to return to its normally open position so that excess GOX can flow through Check Valve A79 and out the vent stack.

GOX Manifold A314 in the main tank sensing panel provides a common ullage pressure source for monitoring conditions within Main Tank A300. Pressure Switch A448 sends a pressurize complete signal to the LOX control panel in the LCC when ullage pressure reaches 20 psig. Pressure Gage A313 displays the ullage pressure. Δ P Transducer A438 signals a liquid level indicator that indicates main tank LOX quantity in gallons on the LCC LOX supply monitor panel. Transducer A437 signals a pressure indicator, also on the LOX supply monitor panel, that displays ullage pressure in psig.

The vent line can be drained of foreign accumulations through Manual Valve A133. Relief Valve A317 and Burst Disc A318 mechanically protect Tank A300 from overpressurization. A line containing Manual Valve A320 allows emergency bypass of Valves A301 and A306. Valve A319 allows bypass of Valve A301. Relief Valves A321 and A322 open at 75 psig to prevent Heat Exchanger A305 and line structural failure.

2.1.5. Replenish Tank Fill. LOX, transported to Complex 37 in trailers, is pumped into Replenish Tank A200 through hoses coupled to the replenish tank coupling-half assemblies (figure 2). LOX passes from the trailers through Coupling-Half Assemblies A123 and A124; through Manual Valves A119 and A66; past Relief Valve A92, which relieves at 75 psig; through Strainer A125; through Manual Valve A211; and into Main Tank A200. When the LOX reaches the desired level, it flows through Trycock Valve A230, which is then closed to terminate the replenish fill operation.

Manual Valve A216 is opened to allow the replenish tank LOX level to be displayed on Differential Pressure Gage A212. The replenish tank fill lines drain through Manual Valves A132 and A69 into a common line and through Check Valve A28. Δ P Transducer A441 signals a liquid level indicator that indicates tank quantity in gallons on the LCC LOX supply monitor.

2.1.6. Replenish Tank Annular Space Evacuation. A portable vacuum pump connected through Manual Valve A228, the vacuum line, and Filter A229 pulls a 20-micron-mercury vacuum within the perlite insulated tank walls. Vacuum within the walls is measured through the line containing Manual Valves A218 and A219 and by Vacuum Gage A217. Relief Valve A237 relieves pressure buildup within the walls at 2.3 psig.

2.1.7. Replenish Tank Pressurization. A vaporization system that converts LOX to GOX pressurizes the replenish tank ullage to 195 psig. This ullage pressure minimizes LOX boiloff losses and force-feeds the LOX to the vehicle containers during replenish operations. Closure of Valves A210 and A211 precedes replenish tank pressurization; then Manual Valve A227 is opened to allow LOX flow to Heat Exchanger A205 for replenish tank pressurization. Dc power on Solenoid Valves A2752 and A2753 allows 750-psig GN₂ control pressure to close normally open Replenish Tank Vent Valve A4. (Manual Valve A220 provides a means of preventing replenish tank pressure leakage should Valve A4 fail to close.) Power on Valves A2754 and A2755 allows the control pressure to open normally closed Replenish Tank Pressurization Valve A201. LOX flows from the tank through Manual Valve A227, Valve A201, and Flow Control Valve A206 which is held in the fully open position by 25-psig GN₂ control

pressure. From Valve A206, LOX flows through Heat Exchanger A205 where it is converted to GOX by ambient air blown across the heat exchanger coils by two motor-operated fans. The GOX passes through the return line to pressurize the tank ullage. A line leading to Controller A235 transmits ullage pressure, which varies Controller A235 outlet pressure from 3 to 15 psig, thus varying the opening through Valve A206 proportionally. In this way a relatively constant ullage pressure is maintained. Should the ullage pressure exceed 195 psig, Pressure Switch A449 in the replenish tank sensing panel would actuate, taking power off Valves A2752 and A2753. This allows Valve A4 to return to its normally open position so that excess GOX can flow through Check Valve A78 and out the vent stack.

GOX Manifold A214, in the replenish tank sensing panel, provides a common ullage pressure source for monitoring conditions within the replenish tank. Pressure Switch A442 sends a pressurize complete signal to the LOX control panel in the LCC when ullage pressure reaches 160 psig. Pressure Gage A213 displays the ullage pressure. Transducer A443 signals an ullage pressure indicator on the LOX supply monitor panel in the LCC.

The vent line can be drained of foreign accumulations through Manual Valve A81. Relief Valve A221 and Burst Disc A222 mechanically protect the replenish tank from overpressurization. A line containing Manual Valve A224 allows emergency bypass of Valves A201 and A206. Valve A223 allows bypass of Valve A201. Relief Valves A225 and A226 open at 300 psig to prevent Heat Exchanger A205 and line structural failure. If a leak should develop between the replenish tank ullage and the hollow manhole cover, Relief Valve A238 would relieve the manhole cover pressure.

2.1.8. Replenish Tank Refill. Should a launch delay occur, Main Tank A300 can refill Replenish Tank A200 through the replenish tank transfer line, which contains Local Transfer Valve A231. Refill operation follows the route from the main tank, through Manual Valve A309, past Relief Valve A90 which cracks at 75 psig to relieve GOX pressure buildup, and into the local transfer line. From the local transfer valve switch in the auxiliary components panel, dc power actuates Valves A2781 and A2782, allowing GN₂ control pressure to open Valve A231. LOX flows through Valve A231 and Valve A227 into the replenish tank. The replenish tank quantity indicator in the LCC LOX supply monitor panel, which receives a signal from Δ P Transducer A441, is monitored to determine when the desired Tank A200 level has been reached.

2.2. Storage Facility, Launcher, LOX Pit, and S-I Stage

2.2.1. Fill and Drain Lines Precool. The S-I precool operation, which prevents overpressurization and choking of the S-I fill line with LOX vapor, requires about 6.5 minutes. Prior to fill and drain line precool operation, dc power actuates Solenoid Valves A2748 and A2749, allowing GN₂ to close S-I Line Vent Valve A16 (figure 3). GN₂ from Manual Valve A5177 in Valve Panel 10 closes S-I Mast Drain Valve A34 through energized Solenoid Valves A2766 and A2767 and opens S-I Main Fill Valve A31 through energized Solenoid Valves A2764 and A2765 (figure 2). GN₂ from the LOX pneumatic control console opens S-I Pump Discharge Valve A7 (figure 3) through energized Solenoid Valves

A2746 and A2747. GN₂ from Valve Panel 10 (through Manual Valve A5177, open Solenoid Valve A5616, and Coupling Halves A6601 and B154) opens Vehicle Fill and Drain Valve B152 (figure 1). LOX flows from Main Tank A300 through Valve A307 and Strainer A99. Manual Valve A112 remains closed during precool operation. The 30-psig GOX ullage pressure provides flow impetus. LOX continues through inoperative Transfer Pump A105, S-I Pump Discharge Valve A7, Check Valve A115, and Manual Valve A26. Relief Valves A88, A110, A139, A140, A141, and A142 protect the fill line and its components. Flow continues through S-I Main Fill Valve A31, Coupling Assembly A4600, Nozzle Assembly B153, Fill and Drain Valve B152, and into the 0-3 sump.

2.2.2. Fill. Main fill of the S-I stage LOX containers commences approximately 6 hours before launch. Prior to containers fill, LOX Vent Valve B163 and No. 1 LOX Vent and Relief Valve B162-1 (pressure operated through Solenoid Valve B215) open to vent the containers during the fill operation. No. 2 LOX Vent and Relief Valve B162-2 (pressure operated through Solenoid Valve B222) also opens for the same purpose. However, Valve B222 allows the vent function of No. 2 Valve B162-2 to operate only prior to launch since a GSE signal is required to actuate Valve B222. When Valves B215 and B222 close, GN₂ trapped between Valves B163, B162-1, and B162-2 and Valves B215 and B222 escapes through the vent ports of Valves B215 and B222 and through Orifices B169.

At the completion of S-I precool operations, Pump A105 starts, drawing LOX from Tank A300, through Valve A307, Strainer A99, through Pump A105, and past Pressure Switch A435, which indicates 'Main Fill in Operation' at the LOX control panel when pump discharge pressure reaches 160 psig. Flow continues through Valve A7, Check Valve A115, Manual Valve A26, Valve A31, Coupling Assembly A4600, Nozzle Assembly B153, Fill and Drain Valve B152, and into the 0-3 sump and to the 0-C sump through the sumps interconnect line. From the 0-C sump, LOX flows to 0-2, 0-1, and 0-4 through the sumps interconnect lines. The containers fill until the operation is terminated by the LOX tanking computer.

The S-I fill and drain line empties through S-I Mast Drain Valve A34 and Check Valves A85 and A150. A GN₂ purge to the motors of Pumps A105 and A106, through Orifices A2769 and A2770, prevents internal explosive atmospheres in the motors and warms the valve and motor components.

2.2.3. Replenish Line Precool. When the containers reach the 75-percent-full level during main fill, a command signal from LOX Tanking Computer A83 initiates the replenish line precool operation. The S-I main fill operation continues during precool. Prior to replenish line precool, the following valves actuate:

- a. Solenoid Valves A2762 and A2763, allowing GN₂ to close Replenish Line Drain Valve A143.
- b. Replenish Valve B151, by GN₂ pressure from Valve Panel 10 Manual Valve A5177 through energized Solenoid Valve A5618 and Coupling Halves A6604 and B160. Solenoid Valve B216 also

actuates through a GSE signal, allowing GN₂ opening pressure to vent through the vent port of Valve B216.

- c. Solenoid Valves A2756 and A2757, allowing GN₂ to open Replenish Line Control Valve A61. Manual Valve A75 is opened.
- d. Solenoid Valves A2760 and A2761, allowing GN₂ to open S-I Throttle Bypass Valve A52.
- e. Replenish and Throttle Control Valve A55 to the fully open position, by 25-psig GN₂ from Valve Panel 5 Regulator A2109.
- f. Solenoid Valves A2758 and A2759, allowing GN₂ control pressure to close Replenish Line Vent Valve A136.

Pressurized LOX flows from Replenish Tank A200 through Manual Valve A210, Strainer A102, Replenish Line Control Valve A61, and Manual Valve A75. Flow continues through Valves A55 and A52, Coupling Assembly A430, Nozzle Assembly B150, and into the 0-4 sump. GOX and LOX vapor relief is provided by Relief Valves A84, A87, A95, A146 and A147. When the LOX level reaches 95 percent, Tanking Computer A83 terminates the replenish precool operation.

2.2.4. Replenish. LOX Tanking Computer A83 controls replenish of the S-I stage LOX containers. This computer in the AGCS generates an electrical analog of sensed differential pressure. The pressure reaches the computer from the vehicle 0-C container pressure sensing lines through Coupling Halves B156, B157, A6605 and A6602. During main fill and replenish operations, a purge of the bottom sensing line is maintained to keep LOX or GOX from reaching the computer. Solenoid Valve A6030 controls this purge by allowing 450-psig helium to flow from Regulator A5280 in Valve Panel 10, through Orifice A6070, Orifice A6071, and Valve A6030. Pressure Switch A6031 actuates at 45 psig, illuminating the LCC auxiliary components panel 'purge pressure on' indicator; at 100 psig the switch deactuates, illuminating the 'purge pressure off' indicator on the same panel, and closes Valve A6030.

The 195-psig GOX ullage pressure in the replenish tank maintains pressurized flow from the tank to the vehicle. The LOX replenish flow route is identical to the replenish precool flow route (paragraph 2.2.3.). When the container LOX level passes 100 percent, ΔP Switch B159 signals the LCC LOX components panel, which displays an overfill indication. (Manual Valve B158 provides a calibration port for Switch B159.) Computer A83 in conjunction with Replenish and Throttle Control Valve A55 maintains the containers at the full level until the automatic sequence begins 150 seconds before ignition. Throttle Bypass Valve A52 remains open until the 98-percent-full level is reached. Replenish and Throttle Control Valve A55 functions similarly to Flow Control Valve A206. The 75-percent-full signal puts dc power on the LOX replenish power bus in the LCC, which then energizes Solenoid Valve A2768, thus allowing GN₂ flow to Controller A58. LOX Tanking Computer A83 positions the Controller A58 poppet in accordance with sensed differential pressures from the vehicle LOX sensing lines.

Upon S-I replenish termination, S-I Stage Replenish Valve B151 closes and Replenish Line Drain Valve A143 opens, allowing LOX to escape from the replenish line through Check Valves A96 and A150.

2.2.5. Drain. Since the S-I and S-IV drain operations function independently, either can commence at any time. The manually actuated S-I drain sequence switch on the LCC LOX control panel actuates the five-minute drain sequence timer, which starts the S-I drain operation. Mast Drain Valve A34 closes, and Vehicle Vent Valve B163 and Vent and Relief Valves B162-1 and B162-2 open. S-I Main Fill Valve A31 and Vehicle Fill and Drain Valve B152 open. Main Tank Pressurization Valve A301 closes, Replenish Tank Vent Valve A4 opens, and Replenish Tank Pressurization Valve A201 closes. Unless previously completed, the main and replenish tanks depressurize while the timer is counting. At the end of the five-minute time delay, Solenoid Valves A2750 and A2751 are energized, opening S-I Line Drain Valve A19. LOX flows from the vehicle containers through the interconnect lines, Fill and Drain Valve B152, Nozzle Assembly B153, Coupling-Half A4600, Main Fill Valve A31, Manual Valve A26, Drain Valve A19, Check Valve A30, Strainer A104, Manual Valve A310, and into the main tank. Gravity provides flow impetus. Relief Valves A89 and A94 relieve LOX and GOX pressure buildup during LOX drain operations.

2.3. Launcher, LOX Pit, and S-I Stage

2.3.1. LOX Bubbling. Helium bubbling through the LOX suction lines prevents temperature stratification within the suction lines and containers. LOX bubbling commences upon initiation of the automatic sequence, continues for 50 seconds, and terminates prior to LOX container prepressurization. 315-psig helium, from valve panel 10 Regulator A5245 (figure 1), flows through Solenoid Valve A5607 and Coupling Halves A6610 and B450 on short cable mast number 4 to the vehicle LOX bubbling ringline which contains eight branch lines, one leading to each LOX suction line. From the ringline, helium flows through Orifice B451 into the suction line and then bubbles up through the LOX in LOX Prevalves B155, the container sump, and the container. The helium vents through No. 1 and 2 LOX Vent and Relief Valves B162-1 and B162-2 and Vent Valve B163, which remain open until container prepressurization begins. The combined flow rate of helium through the LOX is approximately 560 scfm.

2.3.2. Containers Prepressurization. Completion of LOX bubbling and closure of LOX Replenish Valve B151 initiates helium prepressurization of the LOX containers 100 seconds prior to ignition. Valves B162-1, B162-2, and B163 close prior to prepressurization initiation. The containers receive helium from Manual Valve A5242 in valve panel 10 (figure 1) through Solenoid Valve A6029, Orifice A6069, Coupling Halves A6508 and B385 on short cable mast number 2, Check Valve B387, and the O-C diffuser line that leads to the O-C ullage.

The other LOX containers receive ullage pressure (helium) through the GOX interconnect manifold. When ullage pressure reaches 60.0 psia, Pressure Switch B167 removes power from Solenoid Valve A6029, thus stopping helium flow to the vehicle. Valve A6028 remains closed at this time. Should the ullage pressure reach 67.5 psia, Pressure Switch B166 would signal Solenoid Valve B215

to open, thus opening Valves B162-1 and B163. (Manual Valve B168 provides a calibration port for Switches B166 and B167.) Immediately after ignition, ullage pressure drops rapidly. To make up this pressure loss, both Solenoid Valves A6028 and A6029 are opened by Pressure Switch B167, allowing this pressure to be made up through Orifices A6069 and A6068 before liftoff.

2.4. Storage Facility, S-IV Tower Complex, and S-IV Stage

2.4.1. Fill and Drain Line Precool. Prior to precool operation, the following valves actuate:

- a. Solenoid Valve A4006, (figure 2) allowing GN₂ pressure from Valve Panel B (Volume V) to open LOX Main Fill Valve A4005.
- b. Solenoid Valves E319 and E320, (figure 1) allowing helium control pressure from Regulator E206 to open Fill and Drain Valve E151.
- c. Solenoid Valves E212 and E214, opening LOX Container Vent Valves E153 and E154.
- d. Solenoid Valves A4003 and A4004, (figure 2) opening S-IV Main Fill Precool Valve A148.
- e. Solenoid Valves A2744 and A2745, (figure 3) opening Pump Discharge Valve A10.

Actuation of the S-IV precool fill line switch on the LCC LOX control panel initiates S-IV fill line precool upon completion of main and replenish tanks pressurization. Approximately 5 minutes after precool flow starts, Valve A148 closes. The 5-minute open period of Valve A148 allows GOX pressure in the main line to vent through Check Valve A150. Precool flow follows the route from the main tank, through Valve A309, Strainer A100, non-operative Pump A106, Valve A10, Check Valve A45, and Manual Valve A73. Relief Valves A90, A44, A98, A43, and A47 actuate (open) at 300 psig. After Valve A148 closes, flow continues into the Main Fill and Topping Control Valve Complex A4000, through Valve A4005, Filter A4011, Valve A3151, Coupling Assembly A3160, Nipple Assembly E150, Main Fill Valve E151, and into LOX Container E152. As redundant main LOX fill line overpressure protection, Pressure Switch A3172 opens Main Fill Precool Valve A148. Precool takes about five minutes and transfers approximately 4000 pounds of LOX to the S-IV Stage.

2.4.2. Fill. After five minutes of precool flow, the S-IV precool timer shuts off, starting S-IV fill. Upon completion of precool operation, Valve A148 opens and Pump A106 starts. Full flow to the S-IV container begins within 10 seconds after the pump starts and Valve A148 starts closing 30 seconds after the pump starts; thus the main fill line is purged of GOX pressure for 40 seconds. LOX flow follows the precool route described in paragraph 2.4.1. Pressure Switch A436 indicates adequate pump discharge pressure on the LCC LOX components panel.

The S-IV container fills to the 98-percent level at 1000 gpm. LOX Mass Sensor E156 signals Pump A106 shutoff at the 98-percent level, and Solenoid Valves A2740 and A2741 deenergize to open Line Vent Valve A13, allowing LOX to vent to the main tank through Check Valve A76. Fifteen seconds later, Valves A4005 and E151 close and Umbilical Drain Valve A4023 is opened by GN₂ through Valve A4024. LOX in the main line from the vehicle to Valve A4005 escapes through Valve A4023 and Check Valve A4027. After fill to the 98-percent level terminates, Pressure Switch A3172 controls Main Precool Valve A148 as redundant overpressure protection. The S-IV stage remains on standby until the S-I stage LOX loads to the 95-percent level. Pressure Transducer A4002 monitors LOX pressure in the main fill line during the fill operation, and transmits a signal to the DAC propellant loading control and monitor panel when the main line switch is depressed. The pressure is displayed on a voltmeter located on the panel.

2.4.3. Replenish. Prior to replenish, the following valves actuate:

- a. Solenoid Valves A4005 and A4008, which receive power through a switch on the LCC auxiliary components panel, opening Replenish Precool Valve A149 for 30 seconds.
- b. Solenoid Valve A4022, actuated by the DAC PU system to open Replenish Valve A4021.
- c. Line Vent Valve A4023 (closes), through an LCC propellant loading panel switch. S-IV Replenish Valve A4021 automatically controls S-IV replenishing.

Replenish commences through Manual Valve A210, Strainer A102, Replenish Line Control Valve A61, Manual Valve A75, (past Valve A149 that closed after 30 seconds open time) Replenish Valve A4021, Strainer A4011, Umbilical Line LOX Valve A3151, Coupling Assembly A3160, Nipple Assembly E150, Fill and Drain Valve E151, and into Container E152. Pressure Transducer A4018 monitors replenish line pressure during the replenish operation, and transmits a signal to the DAC propellant loading control and monitor panel when the LOX topping line switch is depressed. The line pressure is displayed on a voltmeter located on the panel. Replenish flow continues at 100 gpm until the S-IV system indicates 99.75 percent full. At this level, Replenish Valve A4021 closes and the PU system on the S-IV stage signals Valve A4021 to maintain the S-IV LOX container level at 99.25 to 99.75 percent. Upon automatic sequence initiation, Container E152 is topped to 100 percent; Valves E151 and A4021 then close.

2.4.4. Drain. Upon 100 percent LOX Container E152 fill completion, the DAC PU system opens Umbilical Drain Valve A4023, allowing drainage of the upstream replenish line through Check Valve A4027.

The S-IV LOX container can be drained any time in relation to draining the S-I containers. However, before the S-IV LOX container can be drained, it must be pressurized (paragraph 2.5.3.) and Main Storage Tank A300 and Replenish Tank A200 must be depressurized.

The S-IV LOX container is pressurized with 43-psia helium through Solenoid Valve E236, inoperative Helium Heater E241, and Orifice E240. Vent Valves E153 and E154 close. After the S-IV LOX container is pressurized, Pressure Switch E281 starts a five-minute timer and Main Fill Valve A4005 and LOX Fill and Drain Valve E151 are opened by the DAC PU system.

The storage and replenish tanks depressurize as follows: return of the sequence switch on the LCC LOX control panel to the OFF position removes power from the fill bus causing Vent Valve A1 and Vent Valve A4 to open, Main Tank Pressurization Valve A301 and Replenish Tank Pressurization Valve A201 to close, and the vaporizer fan motors to stop.

After five minutes, on signal from the S-IV drain sequence timer, Solenoid Valves A2742 and A2743 are energized to open Drain Valve A22. Pressurized LOX flows from the S-IV container through Fill and Drain Valve E151, Nipple E150, Coupling Assembly A3160, Umbilical Line LOX Valve A3151, Filter A4011, Main Fill Valve A4005, Drain Valve A22, Check Valve A30, Strainer A104, Manual Valve A310, and into Main Storage Tank A300.

S-IV LOX drain terminates when the S-IV drain sequence timer shuts off causing Fill and Drain Valve E151, Main Fill Valve A4005, and Drain Valve A22 to close.

2.5. S-IV Tower Complex and S-IV Stage

2.5.1. Cold Helium Fill. Cold Helium Spheres E233, E234, and E235, mounted on the LH₂ container inside wall, provide helium for LOX container pressurization during S-IV stage powered flight (figure 1). When the S-IV LH₂ fill level reaches approximately 70 percent, 3000-psig helium at -360°F flows through Coupling Halves A3157 and E225 from the pneumatic distribution system (Volume V). Helium flows through Check Valves E226 and E227 into Spheres E233, E234, and E235.

Should a launch abort occur, Solenoid Valve E232 would dump all helium in the cold helium system overboard. Relief Valve E231 relieves excessive line pressure. Cold helium supply Pressure Switch E245 indicates cold helium low limit sphere pressure during countdown by providing an input to DAC panels and to the automatic countdown sequencer. The circuit from Switch E245 becomes inoperative at liftoff.

2.5.2. LOX Bubbling. Approximately ten minutes prior to automatic sequence initiation (12.5 minutes before ignition), LOX bubbling commences (figure 1). With the cold helium bubbling switch on the pneumatic system control panel in the cooldown position, Bleed Valve E166 opens, allowing helium flow from valve panel B (Volume V) through Coupling Halves A3157 and E225, open Bleed Valve E166, and Orifice E163. Thermal Switch E162 deactuates at -298°F, breaking a circuit to the pneumatic system control panel. When this circuit opens, the cold helium bubbling activated light on the panel goes out. When this light goes out, the cold helium bubbling switch is placed in the bubble position, applying power to the cold helium bubbling control Solenoid Valve E167. This opens Valve E167, allowing helium to flow through Coupling

Halves E225 and A3157, Valve E167, Filter E168, Orifice E169, and to a ring-line that has a branch leading to each LOX suction line. From the ringline, helium flows through Check Valve E171, Orifice E172, and into the LOX suction line. The helium bubbles upward through the LOX into the container ullage space and out Vent Valves E153 and E154. Cold helium bubbling Pressure Switch E170 actuates at 320 psia, making a circuit to the cold helium bubbling activated light, thus indicating sufficient bubbling pressure on the Pneumatic System Control Panel.

Subcooling of the LOX in the suction lines assures proper LOX temperature at the engine LOX pump inlets after a specified in-flight LOX chilldown period.

2.5.3. Container Prepressurization. Prepressurization begins at the start of the automatic sequence (T -150 seconds). S-IV LOX Container Vent Valves E153 and E154 close through Solenoid Valves E214 and E212 deenergization (figure 1). Five seconds later, Solenoid Valve E213 energizes to ensure and maintain the closure of the LOX container vent valves. Cold helium flows from valve panel B into the vehicle through Quick-Disconnect Coupling Halves A3157 and E225. After entering the vehicle, the helium flows through Check Valve E226, Filter E228, Solenoid Valve E236, Plenum E243, Helium Heater E241 (not in operation), Orifice E240, and into S-IV LOX Container E152. When the pressure in LOX Container E152 reaches 47 psia, Pressure Switch E283 actuates and closes Solenoid Valve E236. Pressure Switch E281 actuates at 44 psia, providing the S-IV LOX tank normal pressurized indication on the Propulsion System Preparation and Control Panel. Pressure Switch E280 actuates at 52 psia to provide a LOX container overpressure indication to the Propulsion System Preparation and Control Panel.

2.5.4. Purges. Just prior to the S-IV fill and drain line precool operation, Solenoid Valve A2563 (Volume V) (figure 2) in valve panel B actuates (is energized), allowing 50-psig GN₂ to purge Umbilical Drain Line Solenoid Valve A3151 through Orifice A3171, LOX Coupling Assembly A3160 through Orifice A3170, and swing arm No. 2 umbilical housing through Orifice A3169. These purges prevent development of explosive atmospheres and reduce the probability of components freezing. The S-IV main LOX line purge commences immediately after closure of S-IV Main Fill Valve E151. 50-psig GN₂ flows through Solenoid Valve A2561 (Volume V), Filter A2562, Check Valve A3165, and into the main LOX line. This purge 'inerts' the atmosphere within the main fill line and prevents an explosive hazard at the umbilical housing disconnect after Umbilical Valve A3151 closes.

A purge of the transducer box and junction box on Main Fill and Topping Control Complex A4000 'inerts' the internal atmospheres of the boxes. GN₂ from valve panel B, through Solenoid Valve A2546 (Volume V) and Filter A2547, flows through Orifice A4031 and into the explosion-proof transducer and junction boxes. Relief Valves A4030 provide overpressure protection of the boxes.

2.6. Flight

During vehicle flight, the S-I and S-IV stage LOX systems deliver LOX to the H-1 and RL10A-3 engines respectively. The sequences explained under the heading "Flight" cover the time period from H-1 engine ignition until RL10A-3 engine cutoff.

2.6.1. S-I Containers Pressurization. LOX converted to GOX within Heat Exchangers B30 pressurizes the LOX containers during S-I powered flight. GOX flows from Heat Exchanger B30 through Check Valve B164 to a GOX manifold that collects GOX from all eight Heat Exchangers B30 and transmits it through GOX Flow Control Regulator B170, through the GOX diffuser line into container O-C, and into the O-C ullage. A control line from the O-C ullage to the actuator on Regulator B170 controls the regulator position. In the fully closed position, the regulator delivers GOX at 17.6 pounds per second. As the ullage pressure goes up, the regulator limits flow proportionally. The regulator maintains a pressure of 50 psia within the ullage of all containers through the GOX interconnect lines that connect each outer LOX container ullage to the O-C ullage. Emergency Pressure Switch B166 actuates at 67.5 psia to open, via Solenoid Valve B215, Vent and Relief Valve B162-1 and Vent Valve B163. The vent function of Vent and Relief Valve B162-2 is inoperative during flight. However, mechanical relief setting on Valves B162-1 and B162-2 provides ullage pressure relief at 55 to 62 psig.

2.6.2. S-I LOX Flow. Each S-I stage outboard LOX container supplies LOX to one inboard and one outboard H-1 engine. After ignition, LOX flows from the 0-1, 0-2, 0-3, and 0-4 sumps, through Prevalves B155 that are held open by 750-psig GN₂ (Volume V) through Solenoid Valve B217. After flowing through Prevalves B155, the LOX continues to the inlet side of the H-1 engine turbopumps (Volume VIII). Since all four container sumps connect to the O-C sump through the LOX interconnect lines during flight, the LOX level in each container closely corresponds to the level in the other four containers. Should one engine fail, the LOX in the container that would normally flow to the dead engine would be divided between the remaining seven engines. Prevalves B155 remain open throughout S-I stage powered flight and close simultaneously with the firing of the engine conax valves (Volume VIII and X). Anti-slosh baffles within each LOX container prevent LOX sloshing that might have an adverse effect on vehicle stability.

2.6.3. S-IV Container Pressurization. During S-I powered flight, the S-IV LOX container remains pressurized by Spheres E233, E234, and E235. When the ullage pressure in Container E152 exceeds 46 psia, Pressure Switch E283 signals Solenoid Valve E236 to close. While Solenoid Valve E236 remains open, helium from Spheres E233, E234, and E235 flows through Filter E230, Regulator E229 which drops the pressure to 250 psig, Valve E236, Plenum E243 which reduces line pressure surges, non-operative Helium Heater E241, Orifice E240, and into the ullage of Container E152. Helium also flows through Solenoid Valve E238, Orifice E239, and through the secondary coil of Heater E241.

Warm helium from Heater E241 maintains LOX container pressurization during S-IV powered flight. At a predetermined time after S-I/S-IV separation (Volume X), the flight sequencer signals Solenoid Valve E318 to open, allowing

pneumatic opening of Helium Heater Propellant Valves E215 and E216. LH₂ and LOX flow through Valves E215 and E216, respectively, and into the helium heater combustion chamber; the heater exhaust gases pass out the exhaust nozzle and through the S-IV heat shield center. Helium Heater Thermal Switch E246 senses the temperature in Heater E241 and signals the control circuit to open or close Propellant Valves E215 and E216 through Solenoid Valves E318 and E317 according to whether the temperature is above or below 110°F. Heater E241 operates whenever the temperature is below 110°F. During heater operation, cold helium flows from Spheres E233, E234, and E235 through the route described in the first part of paragraph 2.6.3. When the cold helium enters the coils of Heater E241, it warms, expands, and flows through Orifice E240 to Container E152. Upon S-IV powered flight initiation, control of Solenoid Valve E236 switches from Pressure Switch E283 to Pressure Switch E237. When the helium heater inlet line pressure increases to 320 psia, Pressure switch E237 actuates and closes Solenoid Valve E236. When Pressure Switch E283 ceases to control Solenoid Valve E236, it controls Solenoid Valve E238. When the pressure in Container E152 exceeds 46 psia, Pressure Switch E283 signals Solenoid Valve E238, cutting off helium flow to the heater secondary coils and reducing the warm helium flow.

2.6.4. S-IV LOX Flow. During S-IV powered flight, LOX is drawn through six individual LOX suction lines to the six RL10A-3 engines mounted on the S-IV stage thrust structure (Volume IX). The suction lines connect like wheel spokes around the LOX Container E152 base. LOX Sensor E156, a capacitor using LOX as its dielectric, signals the onboard PU system, providing consumption of nearly all LOX to reduce the weight penalty at S-IV engine cutoff (Volumes IX and X). Should Container E152 ullage pressure exceed 50 psia during S-IV powered flight, the mechanical relief setting on Vent and Relief Valves E153 and E154 would open these valves, providing overpressure relief.

LIST OF FINDING NUMBERS

* FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A1	1	Valve, Pneumatic	6 in., Ball-type	Hydromatics Inc. Model A132S6	10427400	
A2						
A3						
A4	1	Valve, Pneumatic	4 in., Ball-type	Hydromatics Inc. Model A132P5	10427321	
A5						
A6						
A7	1	Valve, Pneumatic	6 in., Ball-type	Hydromatics Inc. Model A132S5	10427325	
A8						
A9						
A10	1	Valve, Pneumatic	4 in., Ball-type	Hydromatics Inc. Model A132P6	10427324	
A11						
A12						

* Location: A = Ground; B = S-I stage; E = S-IV stage; G = Instrument Unit; H = Payload.

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A13	1	Valve, Pneumatic	4 in., Ball-type	Hydromatics Inc. Model A132P5	10427321	
A14						
A15						
A16	1	Valve, Pneumatic	4 in., Ball-type	Hydromatics Inc. Model A132P5	10427321	
A17						
A18						
A19	1	Valve, Pneumatic	6 in., Ball-type	Hydromatics Inc. Model A132S5	10427325	
A20						
A21						
A22	1	Valve, Pneumatic	6 in., Ball-type	Hydromatics Inc. Model A132S5	10427325	
A23						
A24						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A25						
A26	1	Valve, Manual	8 in., Gate-type	Wm. Powell Co. PN 027209	10427454	
A27						
A28	1	Valve, Check	2 in., Swing-type	Wm. Powell Co. Fig. 70-597	10427339	
A29						
A30	1	Valve, Check	6 in., Swing-type	Wm. Powell Co. PN 026178	10427332	
A31	1	Valve, Pneumatic	6 in., Ball-type	Hydromatics Inc. Model A132S5	10427325	
A32						
A33						
A34	1	Valve, Pneumatic	2 in., Ball-type	Hydromatics Inc. Model A132K4	10427346	
A35 through A42 are not functionally applicable to this system.						
A43	1	Valve, Relief	300 psi Relief, 1½ in.	Gardner Johnson & Co. PN 1192	10427327	

FINDING NUMBER	NO. REQ'D	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A44	1	Valve, Relief	300 psi Relief, 1½ in.	Gardner Johnson & Co. PN 1192	10427327	
A45	1	Valve, Check	4 in., 300 psig	Wm. Powell Co. PN 026178	10427322	
A46						
A47	1	Valve, Relief	300 psi Relief, 1½ in.	Gardner Johnson & Co. PN 1192	10427327	
A48						
A49						
A50						
A51						
A52	1	Valve, Pneumatic	3 in., Ball-type	Hydromatics Inc. Model A132M7	10427347	
A53						
A54						
A55	1	Valve, Pneumatic	25 psig Supply Pressure 3-15 psig Signal Pressure, 2 in.	Annin Company Model 1560 W/Domotor Control	10427336	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A56						
A57						
A58	1	Controller, Pneumatic	25 psig GN ₂ inlet 3-15 psig GN ₂ outlet	Servomechanisms, Inc. PN 812089	10434805	53A35A3
A59						
A60						
A61	1	Valve, Pneumatic	3 in., Ball-type	Hydromatics Inc. Model Al 32M7	10427347	
A62 through A65 are not functionally applicable to this system.						
A66	1	Valve, Manual	Shutoff 3 in.	Wm. Powell Co. PN B-50700	10427341	
A67	1	Valve, Manual	Shutoff 3 in.	Wm. Powell Co. PN B-50700	10427341	
A68	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A69	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A70	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-39844	10427342	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A71	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-39844	10427342	
A72						
A73	1	Valve, Manual	6 in.	Wm. Powell Co. PN 027209	10427453	
A74						
A75	1	Valve, Manual	3 in.	Wm. Powell Co. PN 027201	10427340	
A76	1	Valve, Check	4 in., Swing-type	Wm. Powell Co. PN 026178	10427322	
A77						
A78	1	Valve, Check	4 in., Swing-type	Wm. Powell Co. PN 026178	10427322	
A79	1	Valve, Check	6 in., Swing-type	Wm. Powell Co. PN 026178	10427332	
A80	1	Valve, Relief	75 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427326	
A81	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A82						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A83	1	Computer, LOX Tanking			10434613	54A8A5
A84	1	Valve, Relief	300 psig Relief, 1 in.		Part of 10427331	
A85	1	Valve, Check	2 in.	Wm. Powell Co. Fig. 70-597	10427339	
A86						
A87	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	
A88	1	Valve, Relief	75 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427326	
A89	1	Valve, Relief	75 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427326	
A90	1	Valve, Relief	75 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427326	
A91	1	Valve, Manual	Shutoff, Gate-type, 3 in.	Wm. Powell Co. PN B-50700	10427341	
A92	1	Valve, Relief	75 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427326	
A93	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	
A94	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A95	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	
A96	1	Valve, Check	2 in., Swing-type	Wm. Powell Co. Fig. 70-597	10427339	
A97						
A98	1	Valve, Relief	300 psig Relief, 1-1/2 in.	Ladewig Valve Co. PN 11160-C	10427327	
A99	1	Strainer	2500 gpm, 8 in., 150 Micron	Leslie Company	10427337	
A100	1	Strainer	1000 gpm, 6 in., 150 Micron	Leslie Company	10427323	
A101						
A102	1	Strainer	500 gpm, 3 in., 150 Micron	Leslie Company	10427334	
A103						
A104	1	Strainer	150 Micron 6-in., 5 psig ΔP		10427323	
A105	1	Pump and Motor	Motor: 350 HP @ 3560 RPM Pump: 2500 gpm, 3 Ph, 60 cps, 440V	Byron-Jackson Pump Inc. MOD 6x8x11DSM(HAC)LOX	10427328	51B1
A106	1	Pump and Motor	Motor: 200 HP 1000 gpm	Byron-Jackson Pump Inc., Type SMI-LOX 4x6x12 1/2 HH	10427329	51B2

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A107						
A108	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A109	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A110	1	Valve, Relief	300 psig Relief, 1½ in.	Gardner Johnson & Co. PN 1192	10427327	
A111						
A112	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A113	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A114	1	Valve, Check	2 in.	Wm. Powell Co. Fig. 70-597(F.E.)	10427339	
A115	1	Valve, Check	6 in.	Wm. Powell Co. PN 026178	10427332	
A116	1	Valve, Manual	3 in.	Wm. Powell Co. PN B-50700	10427341	
A117	1	Valve, Manual	3 in.	Wm. Powell Co. PN B-50700	10427341	
A118	1	Valve, Manual	3 in.	Wm. Powell Co. PN B-50700	10427341	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A119	1	Valve, Manual	3 in.	Wm. Powell Co. PN B-50700	10427341	
A120	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A121	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A122	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A123	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A124	1	Coupling-Half Assembly	3 in.	Futurecraft Corp. PN 550043-30-002	10427401	
A125	1	Strainer	500 gpm, 4 in., 150 Micron	Leslie Co.	10427338	
A126	1	Strainer	1000 gpm, 6 in., 150 Micron	Leslie Co.	10427323	
A127	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A128	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A129	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A130	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A131	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A132	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A133	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-49008	10427343	
A134						
A135						
A136	1	Valve, Pneumatic	1 in.	Hydromatics Inc. Model A132F1	10427345	
A137						
A138						
A139	1	Valve, Relief	300 psig Relief, $1\frac{1}{2}$ in.	Gardner Johnson & Co. PN 1192	10427327	
A140	1	Valve, Relief	300 psig Relief, $1\frac{1}{2}$ in.	Gardner Johnson & Co. PN 1192	10427327	
A141	1	Valve, Relief	300 psig Relief, $1\frac{1}{2}$ in.	Gardner Johnson & Co. PN 1192	10427327	
A142	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A143	1	Valve, Pneumatic	2 in.	Hydromatics Inc. Model A132K4	10427346	
A144						
A145						
A146	1	Valve, Relief	1 in., 300 psig		Part of 10427331	
A147	1	Valve, Relief	1 in., 300 psig		Part of 10427331	
A148	1	Valve, Pneumatic	4 in., 300 psig		10427324	
A149	1	Valve, Pneumatic	3 in., 300 psig	Hydromatics Inc. Model A132M7	10427347	
A150	1	Valve, Check	300 psig 4-in., Swing-type		10427322	
A151 through A199 are not functionally applicable to this system.						
A200	1	Tank, Replenish	28,000 Gallons		10427472	
A201	1	Valve, Pneumatic	3 in., Ball-type	Hydromatics Inc. Model A132M7	10427347	
A202						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A203						
A204						
A205	1	Vaporizer Assembly			10427318	
A206	1	Valve, Flow Control	3-15 psig Signal, 25 psig Supply, 2 in.	The Annin Co. Model 1660	10427335	
A207						
A208						
A209						
A210	1	Valve, Manual	3 in.			Part of 10427472
A211	1	Valve, Manual	4 in.			Part of 10427472
A212	1	Gage, Liquid Level	0-29500 Gallons	Barton Instrument Co. Model 200-D.P. Type	10427563	
A213	1	Gage, Pressure	6 in., 0-300 psig	U. S. Gauge PN 5003	10427566	
A214	1	Manifold	COX		10427561-20	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A215						
A216	1	Valve, Manual	Shutoff ½ in.	Wm. Powell Co. PN D-39842	10427568	
A217	1	Gage, Vacuum	0-1000 Microns Hg	Consolidated Vacuum Corp., PN TG-025	10427567	
A218	1	Valve, Manual	Shutoff, 2 in.	Vacuum Research Co.	Part of 10427472	
A219	1	Valve, Manual	Shutoff		Part of 10427472	
A220	1	Valve, Manual	Shutoff, 4 in.		75M05969	
A221	1	Valve, Relief	Cracks @ 205 psig Inlet 3 in., Outlet 4 in.		Part of 10427472	
A222	1	Disc, Burst			Part of 10427472	
A223	1	Valve, Manual	3 in.	Wm. Powell Co. PN 027201	10427340	
A224	1	Valve, Manual	Bypass 3 in.	Wm. Powell Co. PN 027201	10427340	
A225	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	
A226	1	Valve, Relief	300 psig Relief, 1 in.	Ladewig Valve Co. PN 1178-C	10427320	

FINDING NUMBER	NO. REQ'D	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A227	1	Valve, Manual	6 in.		Part of 10427472	
A228	1	Valve, Manual			Part of 10427472	
A229	1	Filter			Part of 10427472	
A230	1	Valve, Manual	1 in.	Wm. Powell Co. PN D-39844	10427342	
A231	1	Valve, Pneumatic	6 in., Butterfly-type		75M06583	
A232						
A233						
A234						
A235	1	Controller, Pressure, Pneumatic	Output 3-15 psig Supply 25 psig Sensing 0-300 psig	Mason-Neilan Model 2704	10427402	
A236						
A237	1	Valve, Relief	2.3 psig Relief		Part of 10427472	
A238	1	Valve, Relief	3/4 in., Cracks @ 2 psig		Part of 10427472	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A239		A239 through A299 are not functionally applicable to this system.				
A300	1	Tank, LOX Storage	125000 Gallons		10427471	
A301	1	Valve, Pneumatic	3 in.	Hydromatics Inc. Model A132M7	10427347	
A302						
A303						
A304						
A305	1	Vaporizer Assembly			10427318	
A306	1	Valve, Flow-Control	Supply 25 psig, Signal 3-15 psig, 2 in.	The Annin Co. PN 28265	10427335	
A307	1	Valve, Manual	Shutoff 8 in.	Wm. Powell Co. PN 027088	Part of 10427471	
A308	1	Valve, Manual	Shutoff 6 in.	Wm. Powell Co. PN B-40084	Part of 10427471	
A309	1	Valve, Manual	Shutoff 6 in.	Wm. Powell Co. PN B-40084	Part of 10427471	
A310	1	Valve, Manual	Shutoff 6 in.	Wm. Powell Co. PN B-40084	Part of 10427471	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A311	1	Valve, Gate, Manual	1 in. Trycock	Wm. Powell Co. PN D-49008	10427343	
A312	1	Gage, Liquid-Level	0-140000 Gallons	Barton Instrument Corp. Model 200-D.P. Type	10427564	
A313	1	Gage, Pressure	0-100 psig	U.S. Gauge Co. PN 5003	10427565	
A314	1	Manifold	GOX		10427560-20	
A315						
A316	1	Valve, Manual	Shutoff $\frac{1}{2}$ in.	Wm. Powell Co. PN D-39842	10427568	
A317	1	Valve, Relief	Cracks @ 45 psig	J. E. Lonergan Co. Model 41W209M	Part of 10427471	
A318	1	Disc, Burst	Bursts at 50 psig		Part of 10427471	
A319	1	Valve, Manual	Bypass 3 in.	Wm. Powell Co. PN B-50700	10427341	
A320	1	Valve, Manual	Bypass 3 in.	Wm. Powell Co. PN B-50700	10427341	
A321	1	Valve, Relief	1 in., Cracks @ 75 psig	Gardner Johnson Co. PN 1191	10427326	
A322	1	Valve, Relief	1 in., Cracks @ 75 psig	Gardner Johnson Co. PN 1191	10427326	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A323	1	Valve, Manual	Shutoff, 4 in.		Part of 10427471	
A324						
A325						
A326						
A327	1	Valve, Manual	6 in.		Part of 10427471	
A328	1	Controller, Pressure Pneumatic	Output 3-15 psig Supply 25 psig Sensing 0-40 psig	Mason-Neilan Model 2707	10427403	
A329						
A330						
A331	1	Regulator, Manual	Inlet 50 psig, GN2 Outlet 2.0 psig		Part of 10427471	
A332	1	Valve, Check	1-1/2 psig GN2	Wm. Powell Co.	Part of 10427471	
A333	1	Valve, Manual	Shutoff, GN2	Air Reduction Co. PN 8110	Part of 10427471	
A334	1	Valve, Manual	Shutoff, GN2	Air Reduction Co. PN 8110	Part of 10427471	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A335	1	Regulator, Preset	Inlet 2000 psig, GN ₂ Outlet 50 psig		Part of 10427471	
A336	1	Regulator, Preset	Inlet 2000 psig, GN ₂ Outlet 50 psig		Part of 10427471	
A337	1	Gage, Pressure			Part of 10427471	
A338	1	Gage, Pressure			Part of 10427471	
A339	1	Cylinder, Storage	GN ₂ 222 cu. ft. @ 2000 psig		Part of 10427471	
A340	1	Cylinder, Storage	GN ₂ 222 cu. ft. @ 2000 psig		Part of 10427471	
A341						
A342	1	Valve, Relief, Vacuum and Pressure	6 in.	Oceco Model T	Part of 10427471	
A343						
A344	1	Gage, Compound	0 to 15 psig, 0 to 30 in. Hg Vacuum	Acco Helicoid	Part of 10427471	
A345 through A429 are not functionally applicable to this system.						
A430	1	Coupling Assembly	Bellows-type			75M03191

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A431						
A432						
A433						
A434						
A435	1	Switch, Pressure	Actuates @ 160 psig		10465302	51A11S1
A436	1	Switch, Pressure	Actuates @ 150 psig		10430024	51A11S4
A437	1	Transducer, Pressure			10465306	51A11MT1
A438	1	Transducer, Differential Pressure			10465305	51A11MT2
A439						
A440						
A441	1	Transducer, Differential Pressure	0-8 psid		10465305	51A12MT2
A442	1	Switch, Pressure	Actuates @ 160 psig		10430024	51A12S2

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A443	1	Transducer, Pressure	0-300 psig		10465306	51A12MT1
A444						
A445						
A446						
A447	1	Switch, Pressure	Actuates @ 30 psig		10465306	51A11S3
A448	1	Switch, Pressure	Actuates @ 20 psig		10465301	51A11S2
A449	1	Switch, Pressure	Actuates @ 190 psig		10465303	51A12S1
A450 through A2699 are not functionally applicable to this system.						
A2700	1	Gage, Pressure	0-10,000 psig	U. S. Gauge Co.	10437648	
A2701	1	Gage, Pressure	0-1500 psig	U. S. Gauge Co.	10437688	
A2702	1	Gage, Pressure	0-300 psig	U. S. Gauge Co.	10437687	
A2703	1	Gage, Pressure	0-60 psig	U. S. Gauge Co.	10437686	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2704	1	Regulator, Pressure	Presets, 3000 - 750 psi, 3/8 in.	Grove Valve & Reg. Co. Model 94X	10437651	
A2705	1	Regulator, Pressure	Presets, 750 - 120 psi, 3/8 in.	Grove Valve & Reg. Co. Model 94X	10437651	
A2706	1	Regulator, Pressure	Manual 120 - 25 psig	Moore Products Co. Model 42H50	10437679	
A2707	1	Filter, Pneumatic	30 Micron	Permanent Filter Corp. PN 10813-10/30	10437650	
A2708	1	Valve, Relief	900 ± 50 psig Relief, 750 psig min. Reseat	Republic Mfg. Co. PN 625B-9-6	10437652	
A2709	1	Valve, Relief	120 ± 10 psig Relief, 100 psig min. Reseat	Republic Mfg. Co. PN 625B-3-6	10437680	
A2710	1	Valve, Relief	35 ± 5 psig Relief, 25 psig min. Reseat	Republic Mfg. Co. PN 625B-2-8	10437681	
A2711	1	Switch, Pressure	Actuates 21.5 ± .5 psig Deactuates 20 psig	Southwestern Ind. PN PS-3700A-4	10437682	51A2S3
A2712	1	Valve, Manual	Shutoff	Robbins Aviation PN SSNA-375A-6T	10437684	
A2713	1	Valve, Manual	Shutoff	Robbins Aviation PN SSNA-375A-6T	10437684	
A2714	1	Valve, Manual	Shutoff	Robbins Aviation PN SSNA-375A-6T	10437684	
A2715	1	Valve, Manual	Shutoff	Robbins Aviation PN SSNA-375A-6T	10437684	

FINDING NUMBER	NO. REQ'D	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2716	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2717	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2718	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2719	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2720	1	Valve, Manual	Shutoff, Button-Operated Poppet	Robbins Aviation PN SSNA-375A-6T	10437684	
A2721	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2722	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2723	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2724	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2725	1	Switch, Pressure	Actuates 600 \pm 20 psig Deactuates 550 psig	Southwestern Industries PN PS-5100A	10437683	51A2S2
A2726	1	Valve, Manual	Shutoff	Robbins Aviation PN SSNA-250-4T	10437685	
A2727						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2728						
A2729						
A2730	1	Valve, Manual	Vent, Button-Operated Poppet	Futurecraft Corp. PN 30205	10437647	
A2731						
A2732						
A2733						
A2734	1	Orifice	0.059 in. Dia.	A.U. Stone & Co. Inc.	10430177	
A2735	1	Valve, Relief	55 + 5 psig Relief, 45 psig Min. Reseat	Republic Mfg. Co. PN 6258-3-8	10430169	
A2736	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A13
A2737	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A14
A2738	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A17
A2739	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A16

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2740	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A31
A2741	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A32
A2742	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A35
A2743	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A34
A2744	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A29
A2745	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A28
A2746	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A19
A2747	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A20
A2748	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A22
A2749	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A23
A2750	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A26
A2751	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A25

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2752	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A37
A2753	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A38
A2754	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A41
A2755	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A40
A2756	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A47
A2757	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A46
A2758	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A43
A2759	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	51A44
A2760	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A7
A2761	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A8
A2762	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A9
A2763	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A10

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A2764	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A16
A2765	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A15
A2766	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A12
A2767	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A13
A2768	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	53A35A2
A2769	1	Orifice	10-12 SCFH Flowrate @ 20.7 psig	A.U. Stone Co. PN P-881-3	10428514	
A2770	1	Orifice	10-12 SCFH Flowrate @ 20.7 psig	A.U. Stone Co. PN P-881-3	10428514	
A2771	1	Valve, Manual	Shutoff, 3000 psig GN ₂ , $\frac{1}{2}$ in.	Robbins Aviation Model SSNG 500A-8T	10428576	
A2772 through A2780 are not functionally applicable to this system.						
A2781	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Mod. MV-74	10437618	51A49
A2782	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Mod. MV-74	10437618	51A50
A2783 through A3150 are not functionally applicable to this system.						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A3151	1	Valve, Electropneumatic	3-in.	B. H. Hadley Inc. PN 11409-1	75M05605	57A12A21
A3152	through A3159	are not functionally applicable to this system.				
A3160	1	Coupling Assembly			75M04852	
A3161	through A3164	are not functionally applicable to this system.				
A3165	1	Valve, Check		Douglas Aircraft PN 3871261-501		
A3166	through A3168	are not functionally applicable to this system.				
A3169	1	Orifice-Fitting	0.040 and 0.125 \pm .010 in. I.D.		75M06686-1	
A3170	1	Orifice-Fitting	0.015 and 0.125 \pm .010 in. I.D.		75M06686-2	
A3171	1	Orifice-Union	0.020 in. ID		75M06713-1	
A3172	1	Switch, Pressure	50-250 psig	Custom Components Switches Inc. PN 603G3*S MOD	75M06690	57A12A23
A3173	through A3999	are not functionally applicable to this system.				
A4000	1	LOX Main Fill and Topping Control Assy.	Sled-mounted	Douglas Aircraft PN 7864564-501		429

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A4001						
A4002	1	Transducer, Pressure	LOX, 0-150 psia	Giannini PN 4512715-15		429PT10
A4003	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	3000A1
A4004	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	3000A2
A4005	1	Valve, Pneumatic	Y-type, 4-in. 1000 gpm LOX	Pacific Valves Inc. PN 13648-E0		
A4006	1	Valve, Solenoid	4-way, 2-position	Southwestern Valve Corp PN 804934-1-2		429NS15
A4007	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	3000A5
A4008	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 202873-113 Model MV-74	10437618	3000A4
A4009						
A4010						
A4011	1	Filter	LOX	Westward Engr & Fab.Co. PN 25041		
A4012 through A4017 are not functionally applicable to this system.						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A4018	1	Transducer, Pressure	LOX, 0-150 psia	Giannini PN 4512715-15		429PT11
A4019						
A4020						
A4021	1	Valve, Pneumatic	Y-type, 2 in., 100 gpm	Pacific Valves Inc. PN 13647-FO		
A4022	1	Valve, Solenoid	4-way, 2-position	Southwestern Valve Corp., PN 804934-1-2		429NS17
A4023	1	Valve, Pneumatic	Y-type, 2 in. Drain	Pacific Valves Inc. PN 13647-FO		
A4024	1	Valve, Solenoid	4-way, 2-position	Southwestern Valve Corp., PN 804934-1-2		429NS16
A4025	1	Valve, Relief		Douglas Aircraft PN 3864299-1		
A4026						
A4027	1	Valve, Check		Douglas Aircraft PN 7721093-1	LC-1	
A4028						
A4029	1	Switch, Pressure	Actuates @ 720 psig Deactuates @ 650 psig	Custom Component Switches PN 695-E-24		429PJ1

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A4030	2	Valve, Relief		James, Pond & Clark PN D559T-2M-4		
A4031	1	Restrictor		Douglas Aircraft PN DR4-0.5		
			A4032 through A4599 are not functionally applicable to this system.			
A4600	1	Retractable Coupling Assembly	Bellows-type	Flexonics Inc. PN 107435	75M00253	
		A4601 through A5606 are not functionally applicable to this system.				
A5607	1	Valve, Solenoid	2-way, 2-position 325 psig GN ₂	Marotta Valve Corp. PN 212783-1	10437739	53A67
		A5608 through A5615 are not functionally applicable to this system.				
A5616	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 204424	10425701	53A26
A5617						
A5618	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 204424	10425701	53A71
		A5619 through A6027 are not functionally applicable to this system.				
A6028	1	Valve, Solenoid	With Manual Override, 2-way, 2-position	Marotta Valve Corp. PN 218914	75M02802	53A69

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A6029	1	Valve, Solenoid	With Manual Override, 2-way, 2-position bypass	Marotta Valve Corp. PN 218914	75M02802	53A70
A6030	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 204424	10425701	
A6031	1	Switch, Pressure	Actuates @ 100 psig Deactuates @ 50 psig	Meletron PN M7141EB-32A-2	75M50728-1	
A6032	through A6067 are not functionally applicable to this system.					
A6068	1	Orifice		A. U. Stone PN H264C-114	75M50184-2	
A6069	1	Orifice	Bypass	A. U. Stone PN H-264C-310	75M50184-3	
A6070	1	Orifice		W. O. Leonard PN 156040-5	75M50727-2	
A6071	1	Orifice		A. U. Stone PN P-881-8	75M04165-8	
A6072	through A6507 are not functionally applicable to this system.					
A6508	1	Socket, Quick-Disconnect	3000 psig	Wiggins PN 6400R107A16	75M02214	
A6509	through A6600 are not functionally applicable to this system.					
A6601	1	Socket, Quick-Disconnect	750 psig GN ₂	Wiggins PN 6300R109A4	75M02210	

FINDING NUMBER	NO. REQ'D	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
A6602	1	Socket, Quick-Disconnect	100 psig	Wiggins PN 6200R78A4	75M02212	
A6603						
A6604	1	Socket, Quick-Disconnect	750 psig GN ₂	Wiggins PN 6300R109A4	75M02210	
A6605	1	Socket, Quick-Disconnect	100 psig	Wiggins PN 6200R78A4	75M02212	
A6606 through A6609		are not functionally applicable to this system.				
A6610	1	Socket, Quick-Disconnect	1/2 in., 325 psig GN ₂	Wiggins PN 6200R72A8	75M02207	
A6611 through B149		are not functionally applicable to this system.				
B150	1	Coupling-Half	LOX Replenishing		20M30050	
B151	1	Valve, Pneumatic	Ball-type	Hydromatics Inc. PN 131K13B	20M30045	
B152	1	Valve, Pneumatic	Shutoff, Ball-type	Parker Aircraft Co. PN F61C0017	20M30042	
B153	1	Nozzle Assembly	LOX Fill		20M30202	
B154	1	Nipple, Quick-Disconnect	1/4 in., 750 psig GN ₂	E. B. Wiggins PN 6105R109A4	20M30390	

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
B155	1	Valve, Pneumatic	Ball-type	Parker Aircraft Co. PN F61C0017M1	20M30042	
B156	1	Nipple, Quick-Disconnect	1/4 in.	E. B. Wiggins PN 6005R78A4	20M30138	
B157	1	Nipple, Quick-Disconnect	1/4 in.	E. B. Wiggins PN 6005R78A4	20M30138	
B158	1	Valve, Needle, Three-way	Calibration Valve	Benton Corp. PN 15600	60C27526	
B159	1	Switch, Pressure, Differential	Actuation $22.5 \pm .45$ psid	Servomechanisms Inc. PN 816105, Type TR2124	20M30144	9A21
B160	1	Nipple, Quick-Disconnect	1/4 in.	E. B. Wiggins PN 6105R109A4	20M30390	
B161						
B162-1	1	Valve, Pneumatic	Vent and Relief Relieves @ 57 +5 -0 Psig		20M30460	
B162-2	1	Valve, Pneumatic	Vent and Relief Relieves @ 57 +5 -0 Psig		20M30460	
B163	1	Valve, Pneumatic	Vent Gate-type	North American Aviation PN 9512-48410-61	20M30122	
B164	8	Valve, Check		Precision Equipment Co. PN 126060-2	20M30046	
B165						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
B166	1	Switch, Pressure	Actuates @ 67.5 +.5 psia Deactuates @ 64.0 psia min.	Southwestern Ind. Inc. PN PS-5704-68	60C27761	11A57
B167	1	Switch, Pressure	Actuates @ 60 +.5 psia Deactuates @ 56.5 psia min.	Southwestern Ind. Inc. PN PS-5704-60	60C27760	11A59
B168	1	Valve, Needle, Three-way	Calibration	Benton Corp. PN 17500	60C27547	
B169	2	Orifice-Union Assembly			20M30033	
B170	1	Valve, Butterfly	Fully closed, 17.6 lb/sec Fully open, 22.4 lb/sec	Parker Aircraft Co. PN 5630089	60C27835	
B171 through B214 are not functionally applicable to this system.						
B215	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 218263	20M30128	11A55
B216	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 218263	20M30128	9A17
B217 through B221 are not functionally applicable to this system.						
B222	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 218263	20M30128	11A53
B223 through B384 are not functionally applicable to this system.						
B385	1	Nipple, Quick-Disconnect	1 in., He	E. B. Wiggins PN 6005R82A16	20M30165	

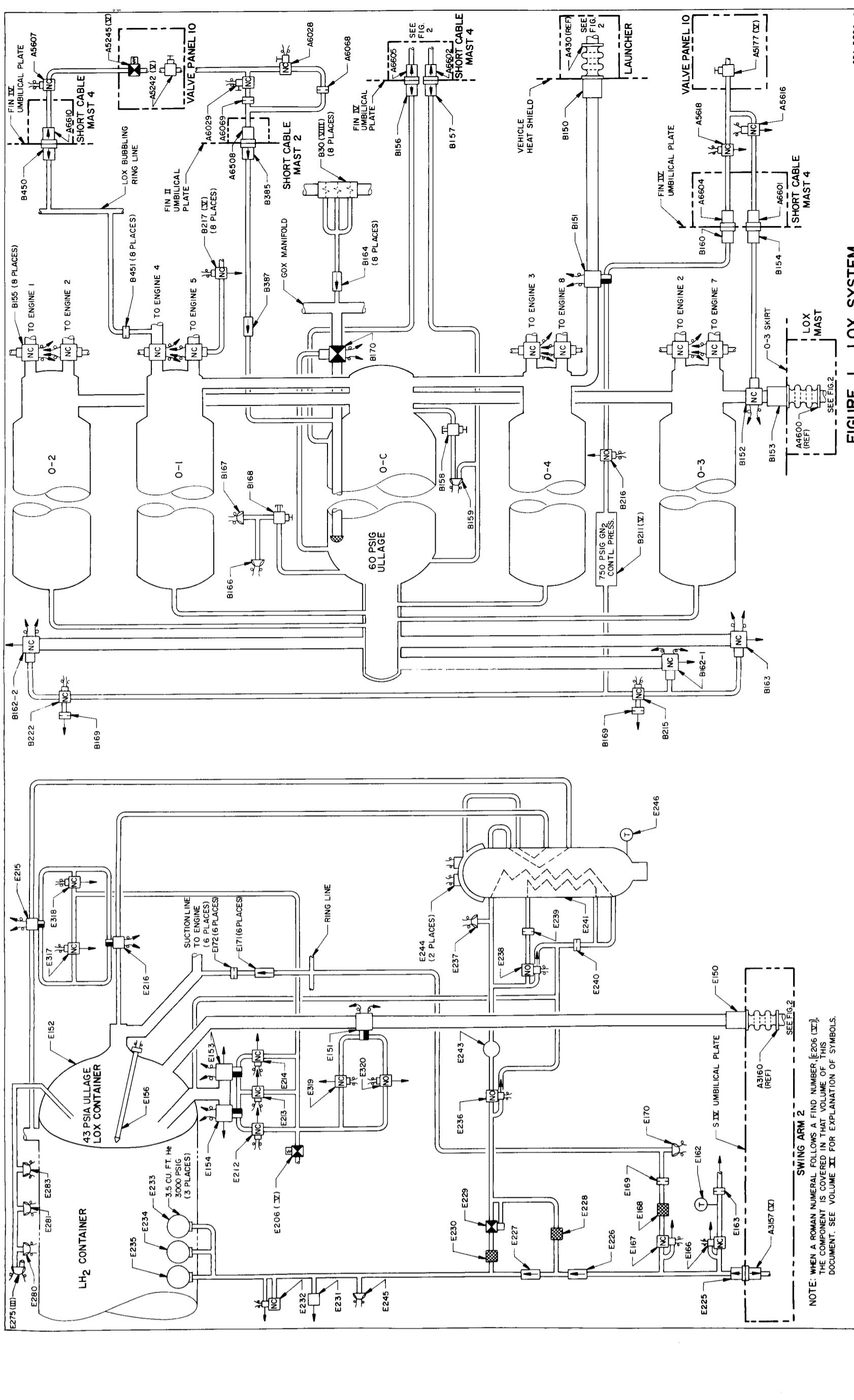
FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
B386						
B387	1	Check Valve Assembly	1 in., He	James, Pond & Clark PN P220T-16BB(L)	20M30379	
B388 through B449	are not functionally applicable to this system.					
B450	1	Nipple, Quick-Disconnect	1/2 in., He	E. B. Wiggins PN 6005R70A8	20M30141	
B451	8	Orifice	.102 Dia, He		20M30199	
B452 through E149	are not functionally applicable to this system.					
E150	1	Quick Disconnect Nipple Assembly		Douglas Aircraft PN 1A84828-1		
E151	1	Valve, Pneumatic		B. H. Hadley PN 11084-11	407A15	
E152	1	Container, LOX	1263 ft ³ 45-48 psia ullage	Douglas Aircraft PN 5863804-507		
E153	1	Valve, Pneumatic	No. 1 Vent and Relief Cracks @ 50 psia Reseat @ 47 psia	Calmec Mfg Corp. PN 230-501	407A11	
E154	1	Valve, Pneumatic	No. 2 Vent and Relief Crack 50 psia Reseat 47 psia	Calmec Mfg. Corp. PN 230-501	407A12	
E155						

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
E156	1	Sensor, LOX		Minneapolis Honeywell PN FG359B-1		408A1
E157	through E161	are not functionally applicable to this system.				
E162	1	Switch, Thermal	Actuates @ $298 \pm 15^{\circ}\text{F}$ Deactuates @ $258 \pm 15^{\circ}\text{F}$	Douglas Aircraft PN 1A65853-1		407A23
E163	1	Orifice	.093 in. Dia.	Douglas Aircraft PN S0268-C6-093		
E164						
E165						
E166	1	Valve, Solenoid		Douglas Aircraft PN 7851845-511		407L13
E167	1	Valve, Solenoid		Douglas Aircraft PN 7851845-511		407L14
E168	1	Filter	10 Micron	Douglas Aircraft PN 7851840-1		
E169	1	Orifice	.031 in. Dia.	Douglas Aircraft PN S0268-C6-031		
E170	1	Switch, Pressure	Pickup 320 ± 10 psia Dropout 115 ± 15 psia	Freibank Co. PN 4940-1		407S12
E171	6	Valve, Check		W. M. Lanagan Co. PN 90048		

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
E172	6	Orifice	.031 in. dia.	Douglas Aircraft PN S0268-C4-031		
E173	through E211	are not functionally applicable to this system.				
E212	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 223544-1	407L9	
E213	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 223544-1	407L10	
E214	1	Valve, Solenoid	3-way, 2-position	Marotta Valve Corp. PN 223544-1	407L8	
E215	1	Valve, Pneumatic	He	Clary Dynamics PN 524122	407A8	
E216	1	Valve, Pneumatic	He	Clary Dynamics PN 524122	407A7	
E217	through E224	are not functionally applicable to this system.				
E225	1	Coupling-Half, Quick-Disconnect	3000 ± 100 psig He	Douglas Aircraft PN 1A84439-1		
E226	1	Valve, Check		W. M. Lanagan Co. PN 90065		
E227	1	Valve, Check		W. M. Lanagan Co. PN 90065		
E228	1	Filter		Aircraft Porous Media Co., AC 4289-61		

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
E229	1	Regulator, Pressure	Inlet 3000 psig He Outlet 250 psig He	B. H. Hadley PN 11089-3		
E230	1	Filter		Aircraft Porous Media Co., PN AC 4289-61		
E231	1	Valve, Relief		Sterer Engineering Co. PN 20500		
E232	1	Valve, Solenoid		Douglas Aircraft PN 7851845-511	407L5	
E233	1	Sphere	3.5 Cu. Ft., 3000 psig He	Douglas Aircraft PN 7851834-501		
E234	1	Sphere	3.5 Cu. Ft., 3000 psig He	Douglas Aircraft PN 7851834-501		
E235	1	Sphere	3.5 Cu. Ft., 3000 psig He	Douglas Aircraft PN 7851834-501		
E236	1	Valve, Solenoid	Pilot-operated	Douglas Aircraft PN 7851845-509	407L6	
E237	1	Switch, Pressure	Pickup 320 \pm 10 psia Dropout 115 \pm 15 psia	Frebank Co. PN 4940-1		
E238	1	Valve, Solenoid	Pilot-operated	Douglas Aircraft PN 7851845-509	407L7	
E239	1	Orifice	0.250 in. Dia.	Douglas Aircraft PN 4851838-C8-250		
E240	1	Orifice	0.375 in. Dia.	Douglas Aircraft PN S4851838D12-375		

FINDING NUMBER	NO. REQD	COMPONENT	REMARKS	VENDOR	DRAWING NUMBER	ELEC SYM
E241	1	Helium Heater Assembly		Douglas Aircraft PN 5851759-503		
E242						
E243	1	Chamber, Plenum		Tavco Inc. PN 23711496		
E244	2	Igniter		General Lab Associates Part of 30104E		
E245	1	Switch, Pressure	Pickup 2940 ± 25 psia Dropout 2840 ± 25 psia	Frebank Co. PN 8008-1	40759	
E246	1	Switch, Thermal	Actuates @ $110 \pm 10^{\circ}\text{F}$	Giannini Controls PN 49849	407A22	
E247 through E279 are not functionally applicable to this system.						
E280	1	Switch, Pressure	Pickup 52.0 ± 1.0 psia Dropout 50.0 ± 1.0 psia	Frebank Co. PN 8015-6	407S8	
E281	1	Switch, Pressure	Pickup 44.0 ± 1.0 psia Dropout 42.0 ± 1.0 psia	Frebank Co. PN 8015-4	407S6	
E282						
E283	1	Switch, Pressure	Pickup 47.0 ± 1.0 psia Dropout 45.0 ± 1.0 psia	Frebank Co. PN 4586-1	407S2	
E284 through E316 are not functionally applicable to this system.						



EIGELBEE | IUY EYEST - SEE PAGE

NOTE: WHEN A ROMAN NUMERAL FOLLOWS A FIND THE COMPONENT IS COVERED IN THAT VOL DOCUMENT. SEE VOLUME **XII** FOR EXPLANATION.

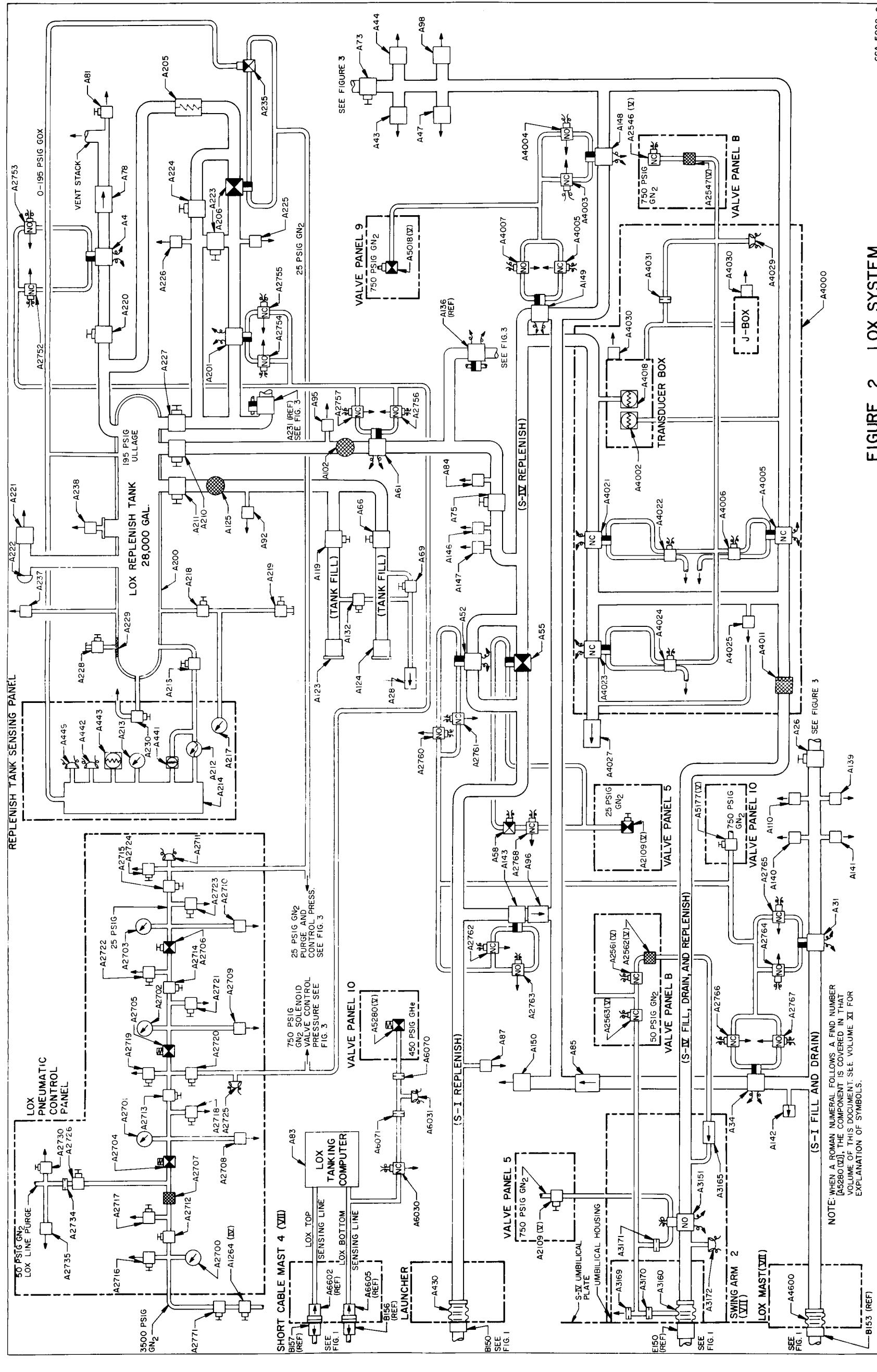


FIGURE 2 IOX SYSTEM

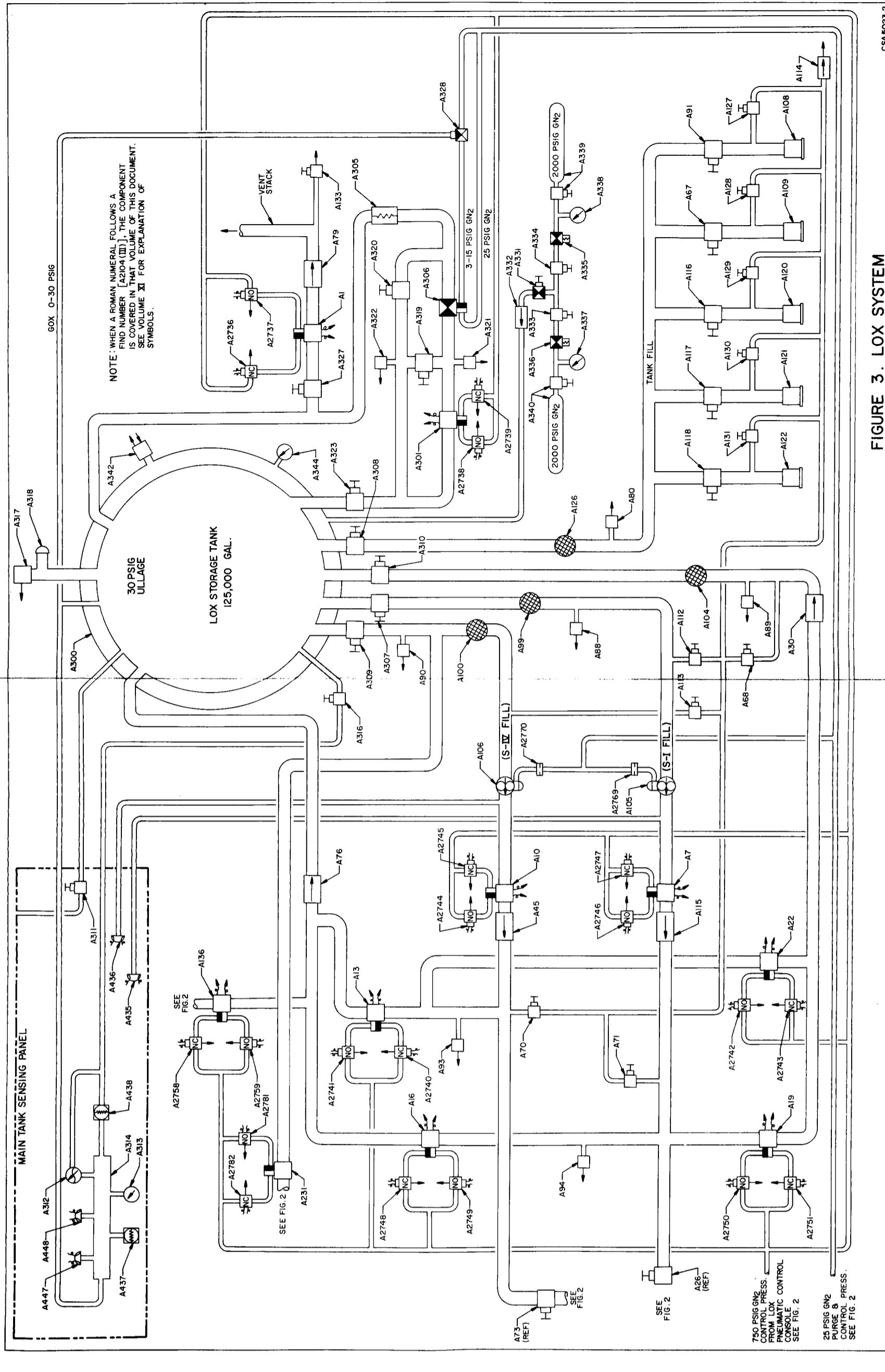


FIGURE 3. LOX SYSTEM

DISTRIBUTION

CCSD-HO (Dept. 4600), BALFOUR, C.L. (10)	K-VG, RIGELL, I. (2)
CCSD-NO (Dept. 2714), BEAMER, W. (14)	K-VG2, GREENFIELD, T.
CCSD-NO (Dept. 2240), SIMMONS, F. (2)	K-VG22, DOWLING, C. (4)
CCSD-NO (Dept. 2752), BECK, R.C.	K-VG23, ROUSE, C. (4)
DAC (Dept. A2-857), BELWOOD, H.	K-VL, GREENE, D. (5)
DAC, KEATING, J.	K-VL, WHISENANT, E. (2)
DAC, HOLLISTER, R. (Dept. A3-770)	K-VM, PICKETT, A.
I-I/IB-SIV, FERGUSON, W.	K-VM2, ROBINSON, G.
I-I/IB-SIVH, LEAGUE, R.	K-VM2, RAINWATER, W.
I-I/IB-SIVL/DAC, STOOPS, G.	K-VM3, PANTOLIANO, T. (3)
I-I/IB-SIVL-NASA/DAC-SACTO, TYSON, O.	K-VM22, DeLaROSA, H.
I-I/IB-SIVL-NASA/DAC-SANTA MONICA, WEAVER, E.	K-VM4, FANNIN, L. (8)
I-MICH-OA, STEVENSON, H.	K-VM23, SCOVILLE, D.
I-MICH-OA, QUINTON, H.	K-VM24, HILL, L.
K-BS27, WHISENANT, R.	LVO-DIR, WILLIAMS, M.
K-DA, POPPEL, T.	LVO-AD, ZEILER, A.
K-DE2, HAHN, R.	LVO-L, BELLAMY, E. (5)
K-DE2, GRIFFIN, F.	R-ASTR-E, FICHTNER, H.
K-DE2, DZIADON, E. (2)	R-ASTR-EA, SMITH, R. (5)
K-DE2, REID, R.	R-ASTR-EAA, PASCHAL, L.
K-DE2, MOORE, R.	R-ASTR-ES, ADEN, R. (3)
K-DE2, STAHLER, S.	R-ASTR-ESI, MILNER, R.
K-DE2, CHAPPLE, E.	R-ASTR-I, HOBERG, O.
K-DE2, HEROLD, C.	R-ASTR-IM, POWELL, J.
K-DE4, DOWNS, J.	R-ASTR-TR, WAGNON, W.
K-DE5, GRIFFIN, F.	R-ME-A, NOWAK, M.
K-DF2, CAREY, T.	R-P&VE-AV, NEIGHBORS, W.
K-DL2, BUCHANAN, D.	R-P&VE/DAC, MEZO, C.
K-DP2, MIMS, W.	R-P&VE-PA, REED, K.
K-DP2, MINTON, C.	R-P&VE-PEM, HOLMES, J. (2)
K-DP2, SPARKMAN, O.	R-P&VE-PM, FUHRMANN, H. (2)
K-DP2, WASILESKI, C.	R-P&VE-PP, HEUSINGER, B.
K-DP4, NELSON, R.	R-P&VE-PT, WOOD, C.
K-DP5, WRIGHT, R.	R-P&VE-V, PALAORO, H.
K-DR2, HOOKER, J. (5)	R-P&VE-VA, HOFFMAN, C.
K-DS232, BUNCH, M.	R-P&VE-VAS, MOON, O.
K-ET, BRIDEWELL, C.	R-P&VE-VF, ROTHE, K.
K-FE, DODD, R. (2)	R-P&VE-VI, FAULKNER, W. (2)
K-AG, RUSSELL, L. (2)	R-P&VE-VK, BOONE, C.
K-GT44, HAWKINS, G.	R-P&VE-VM, BECK, M.
K-L, GORMAN, R.	R-P&VE-VNW, DEVENISH, R. (5)
K-LP2, SWEAT, C.	R-P&VE-VO, KISTLER, W. (20)
K-P, PETRONE, R.	R-P&VE-VS, SCHULZE, W.
K-PA, MATTHEWS, E.	R-P&VE-VSA, PRASTHOFER, W.
K-PC, BODY, R. (2)	R-P&VE-VSI, HURT, W. (2)
K-V, O'HARA, A.	R-P&VE-VSI, KRAUS, G. (2)

DISTRIBUTION (Cont'd.)

R-P&VE-VSP, AKINS, J.
R-QUAL-A, URBANSKI, A. (2)
R-QUAL-AAA, CALDWELL, W.
R-QUAL-J, QUINN, C.
R-QUAL-P, BROOKS, C.

R-QUAL-PI, CONAWAY, J.
R-TEST-CF, FORBIS, M.
R-TEST-I, SIEBER, W.
R-TEST-S, DRISCOLL, D.

Please forward any suggested changes, additions, and corrections
to the Applied Communications Engineering Section (4612), Chrysler
Corporation, Space Division, Huntsville, Alabama.

COPY NO. _____

THIS BOOK CONTAINS _____ PAGES. ONE SIDE ONLY. BACK TO BACK

_____ PAGES HAVE BEEN LEFT INTENTIONALLY BLANK.

THIS BOOK WAS INSPECTED BY R. Fanning AND FOUND TO BE COM-
PLETE.

